# NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS

**TECHNICAL NOTE 3235** 

LOW-SPEED YAWED-ROLLING AND SOME OTHER ELASTIC CHARACTERISTICS OF TWO 56-INCH-DIAMETER,

24-PLY-RATING AIRCRAFT TIRES

By Walter B. Horne, Bertrand H. Stephenson, and Robert F. Smiley

Langley Aeronautical Laboratory
Langley Field, Va.



Washington August 1954

AFMIC

### ud Li Jany Karb, N

NATIONAL ADVISORY COMMITTEE FOR AERONAUTI

#### TECHNICAL NOTE 3235

LOW-SPEED YAWED-ROLLING AND SOME OTHER ELASTIC

CHARACTERISTICS OF TWO 56-INCH-DIAMETER,

24-PLY-RATING AIRCRAFT TIRES

By Walter B. Horne, Bertrand H. Stephenson, and Robert F. Smiley

#### SUMMARY

The low-speed (up to 4 miles per hour) cornering characteristics of two  $56 \times 16$ , type VII, extra-high-pressure, 24-ply-rating tires were determined for a range of vertical loadings, yaw angles, and tire inflation pressures. Locked-wheel drag tests were also made for one vertical-load condition. The quantities measured included cornering force, drag force, self-alining torque, pneumatic caster, vertical tire deflection, rolling radius, and relaxation length. Some supplementary tests were made which included measurements of tire footprint area, vertical-load-deflection characteristics, and the variation of tire radius and width with inflation pressure.

Results indicated that the normal force reached a maximum at between 14° and 18° yaw. The self-alining torque increased with yaw angle up to between 5° and 8° yaw where a maximum was reached. Increasing the yaw angle beyond this point tended to decrease the self-alining torque considerably. The pneumatic caster was a maximum at small yaw angles and tended to decrease in value with increasing yaw angle. The yawed-rolling and sliding drag coefficients of friction both tended to decrease in magnitude with increasing average bearing pressure. In general, the test results indicate that the relaxation length decreases with increasing vertical tire deflection and increasing inflation pressure.

#### INTRODUCTION

Existing experimental data on aircraft tire behavior under static, kinematic, and dynamic conditions, most of which are discussed in reference 1, are limited in both scope and quantity particularly for large tires. This lack of scope has hindered those engaged in design problems concerning landings with yaw, ground handling, and wheel shimmy. A program was therefore undertaken to determine values of the essential tire parameters for a range of tire sizes under static, kinematic, and dynamic

conditions. Some static force-deflection tests of the program have been completed and the results were reported in reference 2. The present paper presents results from part of the kinematic test program for two 56-inch-diameter,  $56 \times 16$ , type VII, extra-high-pressure, 24-ply-rating tires.

Most of the test consisted of towing the tire specimens in a yawed condition. The yaw-angle range covered was from about 0° to 18° and the inflation-pressure range, from about 30 psi to 230 psi. The vertical loads covered ranged from about 10,000 to 45,000 pounds per tire. The towing speed was held constant for each run and did not exceed 4 miles per hour. The quantities measured included vertical tire deflection, side force, drag force, self-alining torque, pneumatic caster, relling radius, and relaxation length. Relaxation-length measurements were also determined for the case of approximately zero yaw for both a standing and rolling tire.

Drag tests were conducted with the wheels locked to obtain measurements in the fore-and-aft direction of the sliding coefficient of friction and the stiffness of the tires. Some supplementary tests were made which included measurements of footprint area, vertical-load—deflection characteristics, and the variation of tire radius and width with inflation pressure.

#### SYMBOLS

Αg gross footprint area, sq in. net footprint area, sq in.  $A_n$ b overall tire-ground contact width, in. đ outside diameter of free tire, in. F force, lb resultant force,  $\sqrt{F_x^2 + F_y^2}$ , 1b  $\mathbf{F}_{\mathbf{R}}$ drag or fore-and-aft force (ground force parallel to direc- $\mathbf{F}_{\mathbf{X}}$ tion of motion), lb cornering force (steady-state ground force perpendicular to  $\mathbf{F}_{\mathbf{y}}$ direction of motion), lb

instantaneous ground force perpendicular to direction of Fy: motion, lb  $\mathbf{F}_{\mathbf{Z}}$ vertical load on tire, lb  $\mathbf{F}_{\mathbf{W}}$ normal force (ground force perpendicular to wheel plane,  $F_y \cos \psi + F_x \sin \psi$ ), lb 2h overall tire-ground contact length, in. fore-and-aft spring constant,  $\left(\frac{dF_X}{dx}\right)_{x\to 0}$ , lb/in.  $K_{\mathbf{x}}$ L relaxation length, in. Lp unyawed-rolling-force relaxation length of tire, in.  $L_{s}$ static relaxation length of tire, in. Ly yawed-rolling relaxation length of tire, in.  $\mathtt{L}_{\lambda}$ unyawed-rolling-deflection relaxation length of tire, in.  $\Delta M_{Z}$ measured portion of self-alining torque, lb-in.  $M_{Z_O}$ self-alining torque for  $\psi = 0.35^{\circ}$ , lb-in. self-alining torque,  $\Delta M_Z + M_{Z_0}$ , lb-in.  $M_{Z}$ N cornering power (rate of change of cornering force or normal force with yaw angle for small yaw angles,  $dF_{\rm v}/d\psi$  or  $dF_{\psi}/d\psi$  for  $\psi \rightarrow 0$ ), lb/degtire inflation pressure, lb/sq in. р tire inflation pressure at zero vertical load  $(F_Z = 0)$ ,  $p_0$ lb/sq in. average gross footprint pressure,  $F_Z/A_g$ , lb/sq in.  $p_g$ 

average tire-ground bearing pressure,  $F_z/A_n$ , lb/sq in.

rated tire inflation pressure, lb/sq in.

 $p_n$ 

 $p_r$ 

g	pneumatic caster, $M_{\rm Z}/F_{\psi}$ , in.
r	outside radius of free tire, in.
$r_{e}$	rolling radius, in.
s	circumferential distance around the tire, ft
v	rolling velocity, ft/sec
w	maximum tire width, in.
x	displacement in direction of motion, in. or ft
δ	vertical tire deflection due to combined vertical and yaw loads, in.
$\delta_{O}$	vertical tire deflection due to vertical load only, in.
η	interpolation factor, in.3/deg
ý	lateral distortion of the tire equator, in.
λο	lateral distortion of tire equator at center of contact, in.
$^{\mu_{X}}$	sliding drag coefficient of friction, $F_{X_{\mbox{max}}}/F_{Z}$
$\mu_{\psi}$	yawed-rolling coefficient of friction, $F_{R_{max}}/F_{z}$
ψ	angle of yaw, deg
ω .	wheel rotation, radians
Subscript	

## max maximum

Bars over symbols denote the average value of the quantities involved for tires A and B.  $\,$ 

5

#### DEFINITIONS OF CONCEPTS

Inasmuch as some of the quantities used in this paper are not generally very well known, the following definitions are given to aid the reader.

Footprint area. The tire contacts the ground in a finite area the shape of which is illustrated in figure 1. Because the tires tested had a rib-type tread, this area consists of alternate strips where the tire contacts the ground and where it does not contact the ground (corresponding to the spaces between the treads). The total area, including the spaces between the treads, is designated as the gross footprint area of the tire. The actual ground-contact area, or bearing area, is referred to as the net footprint area.

Yawed-rolling characteristics.— If a pneumatic-tired wheel is towed unbraked in a straight line while yawed by an angle  $\psi$ , it develops an elastic side force which, for small yaw angles, is roughly proportional to the yaw angle. The component of this side force perpendicular to the direction of motion is called the cornering force  $F_y$ . The initial rate of change of cornering force with yaw angle is called the cornering

power N that is, N = 
$$\left(\frac{dF_y}{d\psi}\right)_{\psi \to 0}$$
. The yawed-rolling condition also

produces a ground moment on the tire which is called the self-alining torque  $M_Z$ . Other yawed-rolling forces are the drag force  $F_X$  (parallel to the direction of motion) and the component of the side force perpendicular to the wheel center plane or normal force  $F_{\psi}$ .

Relaxation length. - The relaxation length is a tire property which has been defined by Temple in reference 3. Consider the situation where the base or ground-contact area of an unyawed tire is laterally deflected relative to the wheel and the wheel is then rolled straight ahead in its initial plane. After the tire has rolled forward a distance equal to the footprint length, then, from that point on, according to the experiments of Kantrowitz (ref. 4) and others, the lateral tire deflection  $\lambda_0$  dies out exponentially with the distance x rolled. The distance that the tire must roll forward in order for the lateral deflection  $\lambda_0$  to drop to a fraction 1/e of its initial value has been defined by Temple as the relaxation length of the tire. In other words, the variation of lateral deflection  $\lambda_0$  with distance x rolled is governed by an equation of the type

$$\lambda_{o} = A_{1}e^{-x/L} \tag{1}$$

where  $A_l$  is a constant depending on the initial lateral tire deflection and L is the relaxation length.

The preceding definition of relaxation length in terms of the result of an unyawed-rolling experiment is only one of a number of ways in which this quantity might be defined. Four other slightly different methods of defining this quantity which were employed in the present investigation will now be described. From the point of view of the theory of reference 5, these definitions and the corresponding experiments should lead to the same numerical value of relaxation length; however, because this theory is not completely rigorous, it is not unexpected that the different definitions to be given will not lead to precisely the same value of relaxation length. In order to distinguish these different values of relaxation length to be discussed, they are assigned different names and symbol subscripts.

Static relaxation length  $L_{\rm S}$ . Consider the experiment where the ground-contact area of a stationary tire is deflected laterally with respect to the wheel plane. The different parts of the center band or equator of the tire are then deflected sidewise from the wheel center plane in the manner illustrated in figure 2. From the available experimental data, it is found that, except in and near the edge of the ground-contact area and near the top of the tire, this distortion curve is essentially an exponential curve of the form

$$\lambda = A_2 e^{-s/L_S} \tag{2}$$

where  $\lambda$  is the lateral distortion of the tire equator, A2 is a constant, and s is the circumferential distance about the tire. The exponential constant  $L_{\rm g}$  is called the static relaxation length of the tire.

Unyawed-rolling-deflection relaxation length  $L_{\lambda}$ . The next definition of relaxation length to be considered is a minor modification of the original definition for the unyawed-rolling case (eq. 1) which is designed to take into account the fact that, owing to experimental difficulties, it was not possible to attain exactly the condition of zero yaw angle in the tests of the present paper. Actually, the minimum yaw angle attained was about 0.35°. In order to take this small yaw angle into account in the definition of relaxation length, the theory of reference 5 indicates that equation (1) should be replaced by the equation

$$\lambda_0 = \lambda_\infty + A_3 e^{-x/L_\lambda} \tag{3}$$

where  $A_{\overline{J}}$  is a constant depending on the initial tire deflection,  $\lambda_{\infty}$  is the measured tire deflection for steady yawed rolling at the small angle of yaw, and  $L_{\lambda}$  is a constant which will be called the unyawed-rolling-deflection relaxation length.

Unyawed-rolling-force relaxation length  $L_{\underline{f}}$ . In view of the fact that tire lateral force is approximately proportional to tire lateral deflection, the relaxation length for the unyawed-rolling-deflection relaxation-length test could also be determined by measuring the rate of decay of the lateral force  $F_{y_i}$  with distance x rolled. This decay proceeds in accordance with the equation

$$F_{y_{\underline{1}}} = F_{y} + A_{\underline{1}} e^{-x/L_{\underline{1}}}$$
 (4)

analogous to equation (3) where  $F_y$  is the measured side force for steady yawed rolling at the small angle of yaw and  $A_{\downarrow}$  is a constant. The corresponding relaxation length  $L_f$  is called the unyawed-rolling-force relaxation length.

Yawed-rolling relaxation length  $L_y$ . Another definition of relaxation length may be obtained in connection with the experiment where a wheel is set up at an angle of yaw and is then rolled straight ahead. If no skidding occurs, then, according to theory (ref. 5) and to the present test results, after the wheel has rolled forward a distance approximately equal to the footprint length 2h, the tire builds up a side force which exponentially approaches an end-point condition for steady yawed rolling; that is, the side force  $F_{y_i}$  builds up with distance x rolled according to a relation of the form

$$F_{y_{\underline{1}}} = F_{y} - A_{5}e^{-x/L_{y}}$$
 (5)

where the constant A<sub>5</sub> depends on the initial tire distortion and L<sub>y</sub> is called the yawed-rolling relaxation length.

Rolling radius. - The rolling radius of a tire is defined as the ratio of the component of the rolling velocity parallel to the plane of the wheel v cos  $\psi$  to the angular velocity of the wheel about the axle  $\omega$ 

or as the ratio of the component of the rolling displacement parallel to the plane of the wheel to the angular rotation of the wheel.

#### APPARATUS

#### Test Vehicle

The basic test vehicle consisted of the fuselage and wing center section of a surplus cargo airplane; general views of this vehicle are shown in figures 3 to 5. The airplane was towed tail first by a tractor truck at an attitude such that the original airplane shock struts were vertical. This attitude was necessary in order to use the existing landing-gear structure and still keep the tires in a vertical plane at varying angles of yaw. The original yokes and torque links of the landing-gear struts along with the wheel assemblies were replaced by steel wheel housings which held the tires and wheels tested. A rigid truss pinned at four points to the two wheel housings held the wheel housings in a fixed relative position during towing operations. Holes located in the wheel housings at angular intervals of  $3\frac{1}{2}^{\circ}$  permitted the wheel frames to be rotated through a nominal yaw-angle range of 0° to  $24\frac{1}{2}^{\circ}$ . Actual measurements on the completed test rig, however, showed the yaw-angle range to be from 0.35° to  $24.9^{\circ}$ .

The towing loads were taken by two steel cables attached between the wheel housings and the tow truck chassis. At high yaw angles with the heavy-weight condition, an additional truck was attached to the tow truck to provide increased power for towing. The maximum towing force required was approximately 8,000 pounds.

The airplane tail was supported by the original swiveling tail-wheel-strut assembly which was modified so that the tail-wheel assembly rotated about a vertical axis. The axle rested in a slot on the top of the tow-truck support structure. This slot and pin arrangement permitted the entire towing load, with the exception of a relatively small amount of friction force, to be carried by the drag cables.

The weight of the test vehicle acting on the tires was approximately 20,000 pounds in the lightest condition. This weight was varied in increments by the addition of six steel and concrete weight cans (each weighing about 8,000 pounds) which were mounted on the airplane structure as shown in figure 3. Additional weights were also added in the fuselage to obtain the heaviest weight condition of 90,000 pounds.

For the locked-wheel drag-force investigation, the test setup was as shown in figure 5. The drag cables were disconnected from the tow truck and attached to a hydraulic ram which was anchored to several stationary heavy vehicles. The airplane wheels were locked by means of rods passing through the spokes and bearing on the wheel frames. Pressure was supplied to the ram by the electrically driven pump shown in figure 5.

#### Instrumentation

The test vehicle was equipped with instruments for measuring side force, self-alining torque, drag, vertical tire deflection, horizontal translation, and wheel rotation. An explanation of the method by which each of the quantities was measured is given in the following sections. Measurements of these quantities were recorded simultaneously on a 14-channel oscillograph mounted in the test vehicle. This oscillograph was equipped with a 0.01-second timer. A sample oscillograph record for a yawed-rolling run is shown in figure 6.

Side force. The side force acting on the two tires was measured by means of a rigid truss structure which was installed between the two wheel housings and which was equipped with four strain-gage dynamometers. (See fig. 7.)

Self-alining torque. The same dynamometers used for the side-force measurements were used to obtain the self-alining torque (that is, the moment about a vertical axis through the center of the wheel).

Vertical load. - The vertical load on the tires was measured by a portable commercial strain-gage dynamometer kit.

Drag forces. - Two separate strain-gage dynamometers, indicated as 5 and 6 in figure 7, were used for measuring the drag forces.

Vertical tire deflection. Vertical-tire-deflection measurements were obtained during the test runs by use of the wheel and strut assembly shown in figure 4(a). Vertical motion of the small solid swiveling wheel, which represents the large tire deflection, is recorded by means of a variable-resistance slide-wire potentiometer.

Fore-and-aft translation. Fore-and-aft translation of the test vehicle from the initial starting point, during the early stages of each run, was recorded by means of the circular slide-wire potentiometer device which is shown in figure 4(a). This device consists of a drum with a string wound around it, one end of the string being tied to a weight resting on the taxi strip. Translation of the test vehicle thus produces rotation of the drum. This rotation produces a sawtooth-type record (see fig. 6), each sawtooth of which corresponds to a single revolution of the drum.

Horizontal translation during later stages of each run was measured by means of trailing bicycle wheels attached behind the wheel frames. These bicycle wheels are equipped with an electrical device which gives a pulse on the oscillograph for each 0.39 foot of distance traveled.

Wheel rotation. Rotation of the test wheels was obtained from revolution-counter devices installed on each of the wheels. These devices furnished indications on the oscillograph for each  $10^{\rm O}$  of wheel rotation.

#### Tires

General description. The tires tested in this investigation were two 56-inch-diameter,  $56 \times 16$ , type VII, extra-high-pressure, 24-ply-rating, rib tread tires which were made by different manufacturers. One tire was new and unused. This tire is referred to in this paper as tire A. The other tire, which will be referred to as tire B, was previously subjected to the static tests which are reported in reference 2. The specifications for these tires given in table I were either obtained from reference 6 or by direct measurement after the conclusion of the tests. Figure 8 shows the deflated and inflated cross sections of the tires tested. These cross sections were obtained from plaster casts taken after the conclusion of the tests and, consequently, they show the tire cross section for a worn condition. There appears to be no appreciable difference in the inflated outer profiles for the two tires. However, there is seen to be an appreciable difference in cross-sectional thickness of the two tire treads.

Tire wear. During the course of the present investigation, there was an appreciable progressive change in the cross-sectional shape of the tires due to skidding and working of the tires. Therefore, the chronological order in which the test data was collected is of some importance in the interpretation of the data. This chronological order is indicated in this paper by a series letter which is assigned to all test data. Specifically, series A represents conditions at the beginning of the tests; series B represents conditions for a later period of time, and so forth.

The change in tire-tread pattern due to tire wear throughout the tests is illustrated in figure 9. At the beginning of the tests both tires had a rectangular cross-sectional tread pattern (fig. 9(a)) and this pattern was substantially preserved throughout most of series A to C. Toward the end of series C, however, the side of the tread in intimate contact with the ground began to wear away and produced the tread pattern illustrated in figure 9(b). This wear increased substantially during series D and for series E to G the tire profile remained approximately as indicated in figure 9(c). The small projecting edges

NACA TN 3235 . 11

of the tread were then cut off and for the remainder of the tests (series H and I) the tread cross section was again essentially rectangular with rounded-off corners, as shown in figure 9(d).

Tire radius. - The variation of the unloaded tire radii with inflation pressure and tire wear is shown in figure 10. Each measurement shown was taken after the tires had been left at constant pressure for at least 24 hours. For test series E to H, the radius of the tires is defined as the maximum radius less the height of the small projecting edge shown in figure 9(c); thus, the indicated difference in tire radius for series A to C and E to I is largely due to the wearing off of the tread. It should also be noted that the tire radii during the later stages of this investigation (series E to I in fig. 10) differ slightly from the radii measured after the conclusion of the tests (table I and fig. 8). The difference, approximately 1 percent, is probably due to the fact that the earlier measurements were made during a period of time when the tires were being regularly subjected to severe loadings whereas the later measurements were made after the tires had been completely unloaded for a long period of time.

A radius-pressure hysteresis loop for tire B is shown in figure ll. The elapsed time from start is shown for a few of the measurements presented. The variation in tire radius for a given pressure is seen to amount to as much as 1 percent for this relatively slow rate of change of pressure (roughly, four hours for most of the cycle).

Tire width. The variation of maximum tire width with inflation pressure is shown in figure 12 for both tires. These measurements were all made after the conclusion of testing (tires well worn) and each measurement was taken after the tires had been kept at constant pressure for at least 24 hours in order to minimize hysteresis effects.

#### Test Surface

All yawed-rolling and drag tests were conducted by towing the test vehicle along the center of a 9-inch-thick reinforced-concrete taxi strip. This taxi strip had a slight crown such that the tires on the test vehicle were subject to a slight tilt. However, this tilt was less than 1°. The texture of the taxi strip, a boarded concrete surface, as determined from plaster casts, is shown in figure 13 for three random positions on the strip. All other tests were conducted on a much smoother, level, reinforced-concrete surface.

#### Precision of Data

The instruments used in these tests and the methods of reducing data are believed to yield results usually accurate within the following limits:

Vertical load on tires, $F_Z$ , percent	•	•	•	•		•	•	•	•	±3
Cornering force, $F_V$ , percent				•	•		•			±3
Force perpendicular to wheel plane, $F_{\psi}$ , percent			•	•	•		•	•	•	±3
Drag force per tire, $F_X$ , lb				•		•	•	•		±300
Measured moment, $\Delta M_Z$ , lb-in		•	•	•	•	•	•	•	•	±3,000
Tire inflation pressure, po or p, lb/sq in	•	•	•	•	•	•	•	•	•	±3
Free radius, r, in				•	•	•	•	•	•	±0.02
Rolling radius, re, in	•	•	•	•	•	•	•	•	•	±0.1
Horizontal translation, x, in										
Vertical tire deflection, $\delta_0$ or $\delta$ , in	•	•	•	•.	•	•	•	•	•	±0.2
Yaw angle. v. deg										±0.1

#### TEST PROCEDURE

The present investigation of tire characteristics is divided into the following parts: yawed-rolling tests, relaxation-length tests, locked-wheel drag tests, and supplementary measurements.

#### Yawed-Rolling Tests

For each run of the yawed-rolling tests, the test vehicle was moved into towing position on the dry, clean, concrete taxi strip and the wheel housings were rotated and locked at the particular yaw angle desired. The tires were adjusted to the test inflation pressure and the vertical tire deflections noted. The vehicle was then towed a distance of approximately 50 feet within a speed range of from 1 to 4 miles per hour. Figure 4 shows one of the tires during a run at 17.90 yaw.

All test runs at yaw angles of 0.35°, 3.9°, 7.4°, 10.9°, 14.4°, and 17.9° were made with both wheels symmetrically yawed with respect to the longitudinal axis of the test vehicle. Because these particular yaw angles were the only angles attainable on the test vehicle, the only way that intermediate yaw angles could be obtained was by unsymmetrically yawing the two wheels with respect to the longitudinal axis of the test vehicle. This was done to obtain intermediate yaw angles of 2.1° and 5.7° (that is, in order to obtain 2.1°, one wheel was yawed by 0.35° and

the other, by 3.9°). When towed ahead in this unsymmetrically yawed condition, the test vehicle first veers off to the side because of the unsymmetrical forces. After a short run, however, the vehicle runs smoothly with the longitudinal axis of the test vehicle yawed with respect to the direction of motion such that both wheels have the same final intermediate yaw angle with respect to the direction of motion.

In general, at the beginning of each run, the tires could not practically be set in an equilibrium condition but were instead subject to a certain amount of distortion (and thus, forces and moments) which appeared as a consequence of residual stresses left over from the previous runs and from moments resulting from the changing of the yaw angles of the wheels.

The following measurements were recorded continuously from the start of the run: side force, self-alining torque, drag force, vertical tire deflection, wheel rotation, and vehicle translation in the direction of motion.

All runs except one (run 185 of table II) were made with the tires toed out in the direction of motion. For this one particular run, the usual direction of motion was reversed in order to give an indication of the effects of unsymmetrical tread wear. In order to investigate this unsymmetrical wear effect more thoroughly, each of the tires was removed from its wheel housing after test series G and was replaced so that the former outboard side of the tire became the inboard side. Tow tests were then continued in the usual toed-out condition throughout test series H and I.

#### Relaxation-Length Tests

Four types of relaxation length were determined in this investigation (see section entitled "Definitions of Concepts"). The methods used to determine these types of relaxation length were as follows:

Static relaxation length  $L_s$ . The standing tires were given initial lateral deflection by pulling outward, by means of hydraulic rams, plates placed underneath the tire. The lateral distortion of each tire center tread relative to the wheel center plane was then measured for several points around the tire circumference between the footprint edge and a point  $180^{\circ}$  from the center of contact.

Unyawed-rolling-force relaxation length  $L_{1}$ . With the wheel housings positioned close to  $0^{\circ}$  yaw (actually  $0.35^{\circ}$ ), the tires were given initial lateral deflections by pulling out on plates placed underneath (as for

the static relaxation-length tests). The test vehicle was then rolled ahead off the plates for a distance of about 50 feet with the recording oscillograph taking a continuous record of side force and horizontal translation.

Unyawed-rolling-deflection relaxation length  $L_{\lambda}$ . This procedure entailed rubbing the center tread of each tire with chalk so that a trace of each center tread would appear on the concrete surface as the tires were rolled ahead. The chalked tires were first given an initial lateral deflection. The test vehicle was then towed forward about 50 feet and the distance between the tire-center-tread traces was measured at intervals from the start up to the point of constant distance between tread traces.

(The rolling procedure for this relaxation length is the same as that for the unyawed-rolling-force relaxation length. Actually, the same test runs were used to obtain test data for determining both of these relaxation lengths.)

Yawed-rolling relaxation length  $L_y$ . The basic data for the yawed-rolling relaxation length were obtained from the initial (force buildup) phase of the 3.90 yawed-rolling tests. This constant was evaluated in this paper for this yaw angle only, since, for larger angles, skidding appeared to be too significant and for smaller angles the accuracy of measurement was usually too poor.

#### Locked-Wheel Drag Tests

The method used to determine tire stiffness and sliding drag in the fore-and-aft direction was as follows: The setup for the locked-wheel drag tests previously described was used, the test vehicle being pulled forward on the taxi strip with the wheels set at approximately  $0^{\circ}$  yaw ( $\psi = 0.35^{\circ}$ ) and locked to prevent wheel rotation. As the test vehicle was pulled forward at a speed less than 10 inches per minute, a continuous record was taken of drag force and horizontal displacement. This procedure was repeated throughout a range of tire inflation pressures for one weight condition. In addition, several runs were made with the concrete surface in a wet condition.

During these tests, the weight of the test vehicle remained constant; however, the vertical load on the tires decreased slightly with increasing drag force as a consequence of the moment produced by the drag force. This change in vertical load was taken into account in the computation of friction coefficients. (It was not taken into account in the other tests since the effect was very small for those conditions.)

#### Supplementary Tests and Measurements

In addition to the tests just described, several supplementary tests were made. These tests included vertical-load-deflection measurements and footprint-area determinations. The vertical-load-deflection characteristics of the two tire specimens were determined for one inflation pressure (about 220 lb/sq in.) with the tires mounted on the test vehicle. Tire-contact or footprint-area measurements were made for the tire specimens at several inflation pressures and vertical tire deflections. These measurements were obtained from the imprint left on a piece of heavy paper placed between a chalked portion of the tires and a smooth concrete hangar floor. Several typical imprints are shown in figure 1.

#### RESULTS AND DISCUSSION

Most of the experimental data obtained from this investigation are presented in tables II to VI and figures 8 to 47.

#### Yawed-Rolling Tests

Table II contains all test data obtained during the final steady-state stage of each yawed-rolling run. Data are presented for 9 different test series (A to I) which represent either different vertical loadings, different tire wear, or different orientation of the tires. The variation of normal force  $\bar{F}_{\psi}$ , self-alining torque  $\bar{M}_{\psi}$ , and pneumatic caster  $\bar{q}$  with yaw angle are shown in figure 14 for all vertical loads and inflation pressures. Some details pertinent to the interpretation of these data are discussed in the appendix. Sample rolling-radius data for two typical test conditions are plotted in figure 15 as functions of yaw angle and vertical tire deflection.

The buildup of cornering force with horizontal distance rolled during the initial stages of the yawed-rolling runs is illustrated in figure 16 for typical runs at several pressures and for three test series. Inasmuch as for most runs there was an initial residual force or preload in the tires, the original test curves did not usually pass through the origin. In order to take this fact into consideration, the test curves shown in this figure have been horizontally shifted (if necessary) so that the extrapolation of each curve is made to pass through the origin. For most of these curves, the initial rate of buildup appears to increase with increasing yaw angle (as is predicted by theory, that is, ref. 5) and that the force generally approaches close to its maximum value before the tires have rolled more than 6 feet.

#### Relaxation-Length Tests

Samples of the test data obtained from the four different methods used to determine the relaxation length of the tire specimens are shown in figure 17 for test series A, B, and E. In parts (a), (b), and (c) of figure 17, these data are plotted in semilogarithmic coordinates in order that the expected exponential curves (according to theory) should appear as straight lines. The relaxation lengths for the conditions shown here and for all other conditions of this investigation were obtained by fitting straight lines to such semilogarithmic plots for each test run and are tabulated together in tables II to IV.

It is seen from figure 17 that usually the test results do appear to give substantially straight lines in these semilogarithmic plots; thus, the theoretical exponential variation of force with distance rolled (for the rolling relaxation lengths) or distance around the tire periphery (for the static relaxation length) is supported. The same data shown for series B in figure 17(b) is replotted in linear coordinates in figure 17(d). The solid lines drawn on these plots (fig. 17(d)) are the same solid faired lines which were fitted through the data in figure 17(b).

It should be noted that, for the static-relaxation-length data, the test data do not agree well with the assumed exponential variation near the two endpoint conditions at the edge of the tire footprint and at the top of the tire (for example, see fig. 17(a)). This discrepancy is due largely to the finite bending stiffness of the tire which requires that the slope of the tire-distortion curve must be zero at the edge of the tire footprint (s = h) and also at the top of the tire and both of these factors conflict with an exponential variation.

#### Locked-Wheel Drag Tests

Most of the experimental data obtained from the locked-wheel drag tests are presented in table V. Also, typical data are shown in figure 18 for the buildup of fore-and-aft force with horizontal distance pulled for several runs.

#### Supplementary Tests

The vertical-load—vertical-tire-deflection characteristics curves for the two tires are given in figures 19 to 21 and the tire-footprint data are given in table VI.

#### Discussion of Parameters

Normal force  $F_{\psi}$ . The variation of the steady-state normal force with yaw angle, obtained from the data in table II, is shown in figure 14 for all test series. For each condition of initial vertical tire deflection  $\bar{\delta}_0$  and inflation pressure  $\bar{p}$ , it is seen that the normal force increases with increasing yaw angle and reaches a maximum value at between 14° and 18° yaw. The maximum normal force tends to decrease with increasing inflation pressure. (This observation is discussed later in more detail under the subject of yawed-rolling coefficient of friction.) For the case of constant inflation pressure, shown in figure 22, increasing the vertical load increases the maximum normal force.

Cornering force  $F_y$ . The cornering force follows substantially the same trends that were described for the normal force, as is shown in figure 23, for a typical loading condition.

Cornering power N.- The variation of cornering power with vertical tire deflection and inflation pressure for the different series tested is shown in figures 24(a) and 24(b), respectively. These data are derived from the initial slope of the normal-force—yaw-angle curves given in figure 14. In order to show more clearly the trends of these data, the effects of inflation pressure and vertical tire deflection, respectively, on cornering power have been isolated in figures 25 and 26. For the constant vertical tire deflection of 3.2 inches considered in figure 25, the cornering power increases nonlinearly with increasing inflation pressure. The data for other constant vertical tire deflections seem to follow this same trend. For the constant-pressure range shown in figure 26, the cornering power increases with increasing vertical tire deflection up to a vertical tire deflection of around 3.0 inches where a maximum is reached. Increasing the vertical tire deflection peyond this point tends to reduce the cornering power.

Self-alining torque  $M_Z$ . The variation of self-alining torque with yaw angle is shown in figure 14 for all test series. The self-alining torque generally increases with increasing yaw angle for yaw angles less than  $5^{\circ}$  to  $8^{\circ}$ . Between  $5^{\circ}$  and  $8^{\circ}$ , a maximum is reached and increasing the yaw angle beyond this range usually decreases the self-alining torque considerably. For constant vertical load, the data indicate that increasing the tire inflation pressure tends to reduce the magnitude of the self-alining torque at most yaw angles. In the case of constant inflation pressure, illustrated in figure 22, increasing the vertical load tends to increase the self-alining torque.

Effect of tire wear on normal force and self-alining torque. The influence of tire wear on normal force and self-alining torque can be seen in figure 27 which presents data both for the unworn tire condition

(series A) and for a considerably worn condition (series F). For the most part, this comparison indicates no extremely large changes in either normal force or self-alining torque; however, reversing the tire tread (reversing the tires in the wheel housings) decreased the maximum normal force considerably as is illustrated in figure 28. A likely explanation for this phenomenon could lie in the fact that the tire-tread beads resulting from unsymmetrical tread wear (see fig. 9) fold over on top of the treads under towing conditions with the tires reversed as is shown schematically in figure 29. Such a folding of the tread bead would tend to reduce the tread-contact area considerably and thus increase the bearing pressure between the tire and the ground. This increase in bearing pressure would result in a reduced friction coefficient (to be discussed later) and would thus reduce the maximum attainable normal force.

Maximum self-alining torque M<sub>Z<sub>max</sub></sub>.— The variation of maximum self-alining torque with inflation pressure is shown in figure 30(a) and with vertical tire deflection by figure 30(b) for the test series investigated. These data are taken from the faired curves drawn through the data presented in figure 14. In order to show more clearly the trends of the maximum self-alining-torque data, the effects of vertical tire deflection and inflation pressure on the maximum self-alining torque have been isolated in figures 31 and 32. For constant inflation pressure (see fig. 31), the maximum self-alining torque tends to increase nonlinearly with increasing vertical tire deflection whereas, for constant vertical tire deflection (see fig. 32), the maximum self-alining torque increases more or less linearly with increasing inflation pressure.

Pneumatic caster  $\bar{q} = \bar{M}_Z/\bar{F}_{\psi}$ . The variation of pneumatic caster with yaw angle for all test conditions is shown in figure 14. The variation of pneumatic caster with vertical tire deflection for the vertical load and inflation pressure range tested is shown in figure 33 for each test yaw angle (data obtained from table II). This figure shows that, for a constant yaw angle, the pneumatic caster increases with increasing vertical tire deflection. For constant vertical tire deflection, any systematic pressure effect on the pneumatic caster is obscured by the scatter of the test data. Faired values of pneumatic caster taken from figure 33 for several constant vertical tire deflections are plotted against yaw angle in figure 34. Figure 34 shows that the pneumatic caster is a maximum at small angles of yaw and generally decreases with increasing yaw angle for the test range covered.

Drag force  $\bar{F}_X$ . The variation of drag force with yaw angle for the rated-vertical-load condition (test series B) for the three different inflation pressures tested is shown in figure 35. The data in figure 35 indicate that the drag force, except at small yaw angles, increases more or less linearly with increasing yaw angle for the yaw-angle range covered.

Essentially the same results were obtained for the other test series and inflation pressures investigated. In order to show trends more clearly, the ratio of drag force to cornering force  $\bar{F}_x/\bar{F}_y$  is plotted against yaw angle in figure 36 for the different test series and tire inflation pressures investigated. If the total horizontal ground force under yawed rolling was normal to the wheel plane, then the drag force  $\bar{F}_x$  would be equal to the cornering force  $\bar{F}_y$  multiplied by the tangent of

the yaw angle or  $\frac{F_X}{\bar{F}_y} = \tan\,\psi.$  Tan  $\psi$  is represented in this figure by

a solid line. In general, the data fall near or above this line and indicate that a small force parallel to the wheel plane usually is present throughout the yaw-angle range investigated.

Yawed-rolling coefficient of friction  $\bar{\mu}_{\psi} = \frac{\bar{F}_{R_{max}}}{\bar{F}_{z}}.$  The variation

of the yawed rolling coefficient of friction with average bearing or ground pressure is shown in figure 37. The data shown in this figure are derived from data given in table II and in figures 19 and 41 (to be discussed later). The trend of the data shown indicates that the yawed-rolling friction coefficient decreases with increasing bearing pressure from approximately 0.94 at 60 pounds per square inch down to approximately 0.56 at an average bearing pressure of 320 pounds per square inch.

Sliding drag (fore-and-aft) coefficient of friction  $\bar{\mu}_{\rm X} = \frac{\bar{F}_{\rm X_{max}}}{\bar{F}_{\rm Z}}$ .

The variation of sliding drag coefficient of friction with average bearing pressure for both dry and wet concrete for the one condition of constant vertical load tested is shown in figure 38. The data shown in this figure was obtained from table V. The sliding drag coefficient of friction for the dry-concrete condition appears to decrease with increasing bearing pressure from approximately 0.85 at a bearing pressure of 50 pounds per square inch down to approximately 0.75 at a bearing pressure of 200 pounds per square inch. The friction coefficients found for the limited number of tests made with the concrete in a wet condition indicate a reduction in the sliding drag coefficient of friction of about 10 to 15 percent over that for the dry-concrete condition.

A comparison of the sliding drag coefficient of friction with the yawed-rolling coefficient of friction is shown in figure 39. Both coefficients seem to show approximately the same trends.

Fore-and-aft spring constant  $\bar{K}_X$ . The variation of fore-and-aft spring constant with tire inflation pressure, obtained from data in table V, for the one constant vertical load condition tested is shown

in figure 40. These data are derived from the initial slope of the fore-and-aft (drag) force  $\tilde{F}_X$  against horizontal displacement  $\tilde{x}$  curves. Samples of these curves for three test inflation pressures are presented in figure 18. For the one constant-vertical-load condition tested, figure 40 indicates that the fore-and-aft spring constant does not change markedly with inflation pressure for the range of conditions investigated.

Vertical-load-vertical-tire-deflection characteristics.- The average vertical-load-vertical-tire-deflection characteristics of the two tire specimens are shown in figure 19 by means of a family of constantvertical-load curves where the ordinate is the initial vertical tire deflection  $\delta_0$  and the abscissa is the tire inflation pressure  $\bar{p}_{\bullet}$ Each constant-vertical-load curve represents one of the test series or constant-vertical-load conditions tested. These curves are faired curves obtained by averaging the data taken from yaw, relaxation length, and footprint area tests. (Test data for yaw angles greater than 2.10 were omitted because, under such yaw angles, large cornering-force preloads were sometimes encountered which could affect the accuracy of the initial vertical tire deflection considerably.) The scatter of the test data was found to range approximately ±0.2 inches and is mainly attributed to hysteresis effects, although tire wear and accuracy of measurement also contribute to the scatter to some extent. Figure 20 shows the hysteresis effects on the test data obtained from the locked-wheel drag tests (table V). The difference between the increasing and decreasing pressure curves is considerable and amounts to as much as 0.5 inch vertical tire deflection. Vertical-load-vertical-tire-deflection data for the one condition of constant tire inflation pressure investigated (p = 219 lb/sq in., unloaded) is shown in figure 21. These data indicate that tire A is slightly stiffer than tire B.

Footprint area  $A_g$ ,  $A_n$ . The variation of gross footprint area  $A_g$ , net footprint area  $A_n$ , and the ratio of net footprint area to gross footprint area  $A_n/A_g$  with vertical tire deflection for the test tires, obtained from the data in table VI, is shown in figure 41. Both the gross footprint area and the net footprint area appear to increase linearly with increasing vertical tire deflection for vertical tire deflections greater than 1 inch. The ratio of net footprint area to gross footprint area appears to decrease slightly with increasing vertical tire deflection and averages approximately 70 percent of the gross footprint area. This ratio will, of course, change for tires having tread designs different from the ones tested.

Footprint length 2h and width b.- The variation of footprint length and width with vertical tire deflection, obtained from data in table VI, is shown in figure 42. These data indicate that both the footprint length and the width increase nonlinearly with increasing vertical tire deflection. Also shown in this figure as solid lines are the

lengths of chords of circles having diameters equal to the free diameter d and maximum width w, respectively, of the tire at its rated inflation pressure and located at a distance  $r-\delta_0$  from the center of the circles. A comparison of these quantities indicates that the experimental values of footprint length and width are smaller than the corresponding chord lengths over the vertical-tire-deflection range tested.

Average bearing pressure  $\bar{p}_n = \overline{F_z/A_n}$  and average gross footprint pressure  $\bar{p}_g = \overline{F_z/A_g}$ . The variation of average bearing pressure and average gross footprint pressure with tire inflation pressure is given in figure 43. The data shown in figure 43 are derived from the faired vertical-tire-deflection—inflation-pressure curves given in figure 19 and from the faired footprint-area—vertical-tire-deflection curves given in figure 41. The heavy solid line shown represents  $\bar{p}_n$  or  $\bar{p}_g = \bar{p}$ . Comparison of this line with the average-bearing-pressure curves indicates that the average bearing pressure becomes increasingly larger than the inflation pressure with increasing inflation pressure. The average gross footprint pressure, on the other hand, decreases as the inflation pressure to being smaller than the inflation pressure at low inflation pressures to being smaller than the inflation pressure at high inflation pressures. The offset of both the average bearing pressure and the gross footprint pressure at zero inflation pressure is indicative of the inherent carcass

Increasing the vertical tire deflection is seen to increase both the average-bearing-pressure and the average-gross-footprint-pressure curves.

stiffness of the tire.

Relaxation length L.- The variation of the four types of relaxation length with tire inflation pressure is shown in figure 44. The variation for the static relaxation length with tire inflation pressure is shown in figure 44(a), for the unyawed-rolling-force and the unyawed-rollingdeflection relaxation lengths in figure 44(b), and for the yawed-rolling relaxation length in figure 44(c). In order to show trends more clearly, the effects of vertical tire deflection on the relaxation length has been isolated in figure 45 where the relaxation length is plotted against tire inflation pressure for several constant vertical tire deflections. For comparative purposes, the same faired lines drawn through the data in figure 45(a) are reproduced in figures 45(b) and 45(c). The agreement between the static and the unyawed-rolling relaxation lengths is fairly good for the vertical-tire-deflection range shown. The yawed-rolling relaxation lengths appear to average about 25 percent less than the static relaxation lengths for the vertical-tire-deflection range shown. Some of this difference is thought to be caused by the twisting moment or self-alining torque present in the yawed-rolling case. As evidence for this fact, some previously unpublished data for a 45-inch diameter tire

(tire C of ref. 2) is presented in figure 46 where the variation of static relaxation length with twisting moment or twist is shown for one condition of combined constant vertical force and constant side force. The moment for this data was applied in a direction such as to tend to simulate the yawed-rolling condition. These data indicate that the static relaxation length decreases with increasing twisting moment or twist.

Figure 47 presents a comparison of static relaxation lengths obtained from reference 2 for a  $56 \times 16$ , 32-ply-rating tire with the data for the  $56 \times 16$ , 24-ply-rating tire specimens used in the present tests. Because the rated inflation pressures differ for these two tire types, a direct comparison of their respective relaxation lengths on the basis of equal inflation pressures would be of dubious significance. Therefore, in order to provide a more significant comparison, the data are shown in figure 47 in terms of the pressure ratio p/pr where pr is the rated inflation pressure for the tire specimen. Fairly good agreement is seen to exist between the different tires for the pressure ratios considered.

In general, the test results indicate that the relaxation length decreases with increasing vertical tire deflection and increasing inflation pressure.

Rolling radius re.- The variation of rolling radius with yaw angle, obtained from data in table II, for two typical test conditions is shown in figure 15(a). The rolling radii for both test tires appear to be in good agreement and remain more or less constant in magnitude with increasing yaw angle up to at least a yaw angle of 14.40. The trend of the data at the higher yaw angles is uncertain because one tire, usually tire A, was observed to be slipping more than the other at this condition. The variation of rolling radius with vertical tire deflection for several of the test inflation pressures is shown in figure 15(b). The data presented in figure 15(b) were obtained from table II and are for yaw angles of 0.350 and 2.10. The trends of these data indicate that, for constant pressure, the rolling radius decreases with increasing vertical tire deflection and that, for constant vertical tire deflection, the rolling radius increases with increasing inflation pressure.

Speed effects. - Within the speed range tested (1 to 4 miles per hour), no apparent effects of speed were found to exist for the quantities measured.

#### CONCLUSIONS

Tow tests were made at the Langley Aeronautical Laboratory primarily to determine the yawed-rolling characteristics of two  $56 \times 16$ , type VII, extra-high-pressure, 24-ply-rating airplane tires. The results of these tests indicated the following conclusions:

- 1. The normal force reached a maximum at between 14° and 18° yaw for the vertical load range tested, and this maximum tended to decrease with increasing tire inflation pressure.
- 2. The cornering power under constant tire inflation pressure increased with increasing vertical tire deflection up to about 3.0 inches vertical tire deflection where a maximum was reached. Increasing the vertical tire deflection beyond this point tended to decrease the cornering power. For the case of constant vertical tire deflection, the cornering power increased with increasing tire inflation pressure.
- 3. The self-alining torque generally increased with increasing yaw angle for yaw angles less than 5° to 8°. Between 5° and 8°, a maximum was reached and increasing the yaw angle beyond this range tended to decrease the self-alining torque considerably.
- 4. The maximum self-alining torque for the case of constant vertical tire deflection increased more or less linearly with increasing tire inflation pressure whereas for the case of constant inflation pressure the maximum self-alining torque increased nonlinearly with increasing vertical tire deflection.
- 5. The pneumatic caster was a maximum at small angles of yaw and generally decreased with increasing yaw angle for the test range covered.
- 6. The yawed-rolling friction coefficient decreased with increasing bearing pressure from approximately 0.94 at a bearing pressure of 60 pounds per square inch down to approximately 0.56 at a bearing pressure of 320 pounds per square inch.
- 7. The sliding drag coefficient of friction for the dry concrete condition also decreased with increasing bearing pressure from approximately 0.85 at a bearing pressure of 50 pounds per square inch down to 0.75 at a bearing pressure of 200 pounds per square inch. The limited number of tests made with the concrete in the wet condition indicated a 10 to 15 percent reduction in the sliding drag friction coefficient over the dry-concrete condition.

8. In general, the test results indicate that the relaxation length decreases with increasing vertical tire deflection and increasing inflation pressure.

Langley Aeronautical Laboratory,
National Advisory Committee for Aeronautics,
Langley Field, Va., June 11, 1954.

\*

------

#### APPENDIX

#### DETAILS OF CORNERING-FORCE AND SELF-ALINING-TORQUE MEASUREMENTS

In conducting the yawed-rolling tests a certain amount of difficulty was encountered as a consequence of the fact that it was impractical to set the wheels on the test vehicle exactly to a zero yaw condition; the minimum attainable angle was about 0.35°. This fact is important because, in order to obtain cornering-force and self-alining-torque data for any yaw angle, it is necessary to subtract the dynamometer-record deflections for the yawed-rolling case from those for the reference zero yaw case. In the case of cornering force, this difficulty was resolved by rolling the test vehicle both forward and backward at the minimum yaw angle (0.35°) and by taking the average record deflection for the two cases as corresponding to the zero yaw condition. The cornering force for this 0.35° yaw angle was taken as half the difference in force for the rolled-forward and rolled-backward conditions.

For the self-alining torque, the record deflections for the zero yaw condition could not be accurately determined; consequently, it was not possible to obtain direct measurements of the self-alining torque. Instead all moment measurements were made with reference to the 0.35° minimum yaw condition; that is, the measured moment (designated as  $\Delta M_Z$ ) represents the difference in self-alining torque for the yaw angle considered and for the 0.350 condition. In order to obtain the actual self-alining torque Mz, it is necessary to add to the measured quantity  $\Delta M_Z$  the self-alining torque for 0.35° yaw (designated as  $M_{ZO}$ ) or  $M_Z = M_{Z_O} + \Delta M_Z$ . The quantity  $M_{Z_O}$  was estimated from the statictorsional-elasticity-test data for one of the test tires which is given in reference 2. This procedure for correcting the measured moment AMZ to Mz is believed to be fairly satisfactory both because the torsional elastic properties of static and slowly rolling tires are at least approximately similar (for example, see experimental data in ref. 5) and because the correction term  $M_{Z_{\mathrm{O}}}$  is usually relatively small in comparison with the measured term  $\Delta M_2$ .

#### REFERENCES

- 1. Hadekel, R.: The Mechanical Characteristics of Pneumatic Tyres. S & T Memo. No. 5/50, British Ministry of Supply, TPA 3/TIB, Mar. 1950.
- 2. Horne, Walter B.: Static Force-Deflection Characteristics of Six Aircraft Tires Under Combined Loading. NACA TN 2926, 1953.
- 3. Temple, G.: Note on American Work on Kinematic and Dynamic Shimmy. Rep. No. A.D. 3158, British R.A.E., Nov. 1940.
- 4. Kantrowitz, Arthur: Stability of Castering Wheels for Aircraft Landing Gears. NACA Rep. 686, 1940.
- 5. Von Schlippe, B., and Dietrich, R.: Zur Mechanik des Luftreifens. Junkers Flugzeug- und Motorenwerke, A.G., Dessau. (Available from CADO as ATI 105296.)
- 6. Anon.: Military Specification Casings; Aircraft Pneumatic Tire.
  Military Specification, MIL-C-5041, Sept. 16, 1949; Amendment-2,
  Feb. 8, 1951.

TABLE I .- TIRE SPECIFICATIONS

Specifications	Military specification	End of te	st, tires ondition
	(ref. 6)	Tire A	Tire E
Tire:			
Type <sup>a</sup>	<u>vii</u>		
Ply rating	24		
Static load, 1b	45,000		
Inflation pressure, lb/sq in	178		
Burst pressure, lb/sq in	712 (minimum)		
Moment of static umbalance, oz-in	90	~~~~	
Diameter, deflated, in		54.40	53.70
Diameter, inflated, in	55.26 (minimum)		
	56.40 (maximum)	56.16	56.24
Maximum width, deflated, in		15.36	15.87
Maximum width, inflated, in	16.00 (maximum)	16.00	16.06
Bead width, in.	3.88 (maximum)	2.80	3.00
Minimum wall thickness, in		1.1	0.9
Wall thickness at tread center line			
(including tread), in		1.3	1.0
Depth of tread (at tread center			
line), in	0.35 (maximum)	0.15	0.15
Casing weight, lb	268 (maximum)	253	247
Tread pattern	Rib or nonskid	Rib	Rib
Innertube:			
Thickness, in		Ċ	.2
Weight, 1b			• 7
Wheel:			
Rim diameter, in.	ì	30	5
		-	17
Weight, lb	· · · · · · · · ·	2	<b>-</b> 1

<sup>&</sup>lt;sup>8</sup>Type VII is an extra high pressure tire.

TABLE II. - YAW TEST DATA (a) SERIES A:  $\tilde{F}_{g}$  = 17,000 lb;  $(F_{g})_{tire\ A}$  = 17,000 lb;  $(F_{g})_{tire\ B}$  = 16,900 lb.

					Average	Value	1						Tire .	A			Tire	В	
Run	ī, lb/sq in.	δ <sub>o</sub> , in.	ā, in.	₹, deg	ř <sub>y</sub> , 1b	<b>j</b> <sub>x</sub> , 1b	Ē <sub>ψ</sub> , 1b	∆M <sub>2</sub> , lb-in.	й <sub>z</sub> , lb-in.	ਸ਼ਿਦਾ	Ľ <sub>y</sub> , in.	p, lb/sq in.	ō <sub>o</sub> , in.	B, in.	r <sub>e</sub> , in.	p, lb/eq in.	ō <sub>o</sub> , in.	ð, in.	re, in.
1 2 5 4 5 6 7 8 9 10 11 12 15 16 17 18 19	626655766657665565665588755	35.77 89.6666 75.77.77.6666 7.9	355 - 355 - 535 - 444 - 22	0.355 .355 .355 .355 .355 .355 .355 .355	620 530 630 530 640 2,980 5,930 4,980 11,680 11,680 11,680 11,540 12,540	* * * 400 600 1,200 1,800 2,500 3,200 3,200 3,100 * 400	620 580 570 690 570 640 2,990 1,940 1,940 11,940 13,720 760 7,850	0 0 0 0 0 0 0 17,500 57,600 57,500 21,600 29,100 13,800 12,500 18,400	4,700 4,700 4,700 4,700 4,900 4,500 21,600 34,700 44,400 34,600 19,500 24,100 24,100 25,500	6.94 8.87 6.81 6.82 7.65 4.74 6.82 7.65 4.74 6.83 7.65 4.74 6.83 6.83 7.65 6.83 7.65 6.83 6.83 7.65 6.83	*****	58 56 55 55 66 57 55 55 66 61 61 59 59 88 88	3.5.7.7.1.8.0.7.7.6.9.5.7.7.6.6.6.6.6.6.6.6.6.6.6.6.6.6.6.6.6	577 18074 088 11556 77	25.5 25.4 25.5 25.5 25.5 25.7 25.7 25.7 25.7 25.7	6516655866665865465666598456	5777 + 98 5 5 6 6 5 7 7 7 7 6 6 6 6 7 8 0	5577 9854 581 2210 76	25.4 25.5 25.5 25.6 25.8 25.8 25.8 25.7 25.7 26.1
120 El 20 El	85 82 84 89	2.8 2.8 2.8 2.7	3.2 3.2 3.2	3.9 7.4 10.9 14.4	3,840 6,780 10,380 12,210 12,580	700 1,500 2,000 3,100	6,810 10,490 12,370 12,960	28,500 27,900 19,700 11,000	32,300 32,100 23,900 15,300 2,400	4.74 3.06 1.93 1.18 3.29	12.8 ** ** **	80 85 94	2.9 2.8 2.8 2.7 2.0	2.9 3.4 3.1 3.1	26.1 26.2 26.2 26.7	83 83 84	2.8 2.8 2.8 2.7	3.1 2.9 3.3 3.2 1.9	26.1 26.2 26.2 26.0
25 26 27 28 29 39 31 32 32 32 33 32 32 32 32 32 32 32 32 32 32 3	169 169 168 164 168 168 170 164	1.9 1.9 1.9 1.9 1.9 1.9	1.9 1.9 1.9 1.8 2.0 2.0	255 255 2.1 3.9 7.4 10.9 14.4	750 800 790 890 4,540 7,640 11,030 12,550	* * 300 400 1,500 2,000 2,900	800 790 890 4,550 7,650 11,130 12,700 12,350	0 0 8,200 19,700 20,700 18,700 7,400	2,400 2,400 2,400 10,600 21,900 25,300 21,300 9,300	3.00 3.04 2.70 2.33 2.86 2.09 1.68	** ** ** 11.9 ** **	175 169 169 167 165 168 169 173 168	1.9 2.0 2.0 2.0 1.9 2.0	1.9 2.0 2.0 1.9 1.8 2.1	27.0 27.0 27.0 26.9 26.9 27.1 26.7 27.3	175 169 169 168 162 168 166 168	1.9 1.9 1.9 1.7 1.8 1.9	1.9 1.9 1.9 1.7 1.8 2.0	27.0 27.0 27.0 27.0 27.0 27.1 27.0 26.7

<sup>\*</sup>Force too small to be measured accurately.
\*\*\*Relaxation length not determined.

TABLE II. - YAW TEST DATA - Continued (b) SERIES B:  $\bar{\mathbf{F}}_z$  = 25,700 lb;  $(\mathbf{F}_z)_{\text{tire A}}$  = 23,700 lb;  $(\mathbf{F}_z)_{\text{tire B}}$  = 25,800 lb.

Rum					A	verage va	lues						Tire	A			Tire	В	
мщ	lb/sq in.	δ <sub>ο</sub> , in.	ā, in.	₹,	f <sub>y</sub> , 1b	ξ <sub>x</sub> , 1b	₽ţ, lb	∆H <sub>z</sub> , lb-in.	M <sub>z</sub> , lb-in.	đ, in.	fy,	p, 16/sq in.	δο, in.	ð, in.	r, in.	p, lb/sq in.	δ <sub>o</sub> ,	8, in.	re,
3355356 338 339 339 339 339 341 341 341 341 341 341 341 341 341 341	12 12 13 14 14 15 15 15 15 15 15 15 15 15 15 15 15 15	5.989989919099 5.555555555555555555555555555555555	333333333333333333333333333333333333333	0.55 .35 .35 .35 .35 .35 .21 2.1 3.9 7.4 10.9 14.4 14.4	780 830 870 930 720 3,280 3,060 5,830 10,880 13,880 16,400 	* * * 600 300 1,000 1,000 3,100 2,900 4,200 - 4,900	780 830 870 970 970 720 3,300 5,880 11,030 14,720 14,180 16,930	0 0 0 0 0 0 23,600 30,400 51,000 56,400 37,200 27,300	5,300 5,500 5,500 5,600 5,400 36,700 60,300 43,200 44,100 34,400 44,100	7.05 6.32 6.32 6.02 7.50 11.69 6.24 5.47 2.93 3.11 2.03	*** *** *** *** *** *** *** ***	79 88 83 88 79 80 79 79 81 78 80 81 127	5.888.889.91.90.98 5.888.89.91.90.98 5.888.89.91.90.98	5.88 5.88 5.88 5.88 7.54 4.66 4.9 2.8	25.76.66.625.76 25.76.25.77 25	79 78 79 79 83 82 79 79 78 79 79 79 79	3.898 3.98 3.998 3.99 3.99 4.00 4.00 4.00 4.00	3.98 3.98 3.98 3.99 3.70 4.25 4.7 4.9 2.8	25 25 25 25 25 25 25 25 25 25 26 25 25 25 26 25 25 25 25 25 25 25 25 25 25 25 25 25
30 H 20 H	128 128 126 126 127 127 128	2.8 2.8 2.8 2.8 2.7 2.7	20.000000000000000000000000000000000000	2.1 2.1 3.9 7.4 10.9 10.9 14.4 17.9	5,240 4,860 8,800 13,600 15,350 15,490 16,330 15,690	\$00 \$00 900 2,200 2,900 3,000 3,800 \$,800	5,250 6,840 13,770 15,620 15,780 16,760	27,500 25,300 30,600 39,000 25,900 22,200 23,400 21,400	51,500 29,400 34,200 43,500 28,200 26,500 27,900 25,700	5.96 6.04 3.87 3.14 1.81 1.68 1.66	17.9 ** ** ** ** **	126 128 126 126 126 128 126 126	2.9 2.8 2.8 2.8 2.7 2.7	2.7 2.8 2.7 3.0 2.9 3.0 2.9	26.5 26.6 26.6 26.5 26.1 26.4 26.6 27.5	126 128 128 126 126 127 127	2.8 2.7 2.8 2.7 2.8 2.7 2.8 2.7 2.8 2.7	2.6 7.4 8 9.9 1.8 2.9 2.8	*******
789012545678901	202 205 205 205 205 205 200 205 200 202 202	2.1 2.1 2.0 2.1 2.1 2.1 2.1 2.1 2.0 2.0 2.0 2.1	2.1 2.1 2.0 2.1 2.1 2.2 2.0 2.1 2.1 2.1 2.2 2.1 2.1 2.2	.35 .35 .35 .35 .35 .35 .35 .35 .35 .35	1,050 1,050 1,210 1,070 990 990 5,070 5,210 9,400 13,700 14,950 15,510 14,950	* * * * * * * * * * * * * * * * * * *	1,050 1,050 1,210 1,070 950 980 5,080 5,220 9,130 15,780 15,180 15,000 15,690	0 0 0 0 0 0 0 0 0 20,500 25,100 17,900 16,500 14,100	3,200 -3,200 3,200 3,200 3,200 3,200 3,200 23,700 24,900 25,600 26,300 21,100 19,700 16,800 21,300	5.05 5.05 2.05 2.05 5.22 7.75 2.05 2.05 2.05 2.05 2.05 2.05 2.05 2.0	** ** ** ** ** **	202 203 202 206 203 203 203 200 205 202 202 202 202 202	2.5 2.1 2.1 2.1 2.1 2.1 2.1 2.2 2.0 2.1	2.5 2.1 2.1 2.1 2.1 2.2 1.8 2.1 2.1 2.1 2.1 2.1 2.1 2.1 2.1 2.1	27.1 27.1 27.2 27.2 27.2 27.2 27.2 27.1 27.2 27.2	800 300 800 800 800 800 800 800 800 800	2.0 2.0 1.9 2.1 2.1 2.1 2.1 2.1 2.1 2.1 2.1	2.0 2.1 2.0 1.9 2.1 2.0 2.5 2.1 2.0 2.5 2.1 2.0 2.5 2.1 2.0 2.5	27. 27. 27. 27. 27. 27. 27. 27. 27. 27.

<sup>\*</sup>Force too small to be measured accurately.
\*\*\*Relexation length not determined.

TABLE II. - YAW TEST DATA - Continued

(c) SERIES C:  $\bar{r}_x = 32,700$  lb;  $(\bar{r}_x)_{\rm tire\ A} = 32,900$  lb;  $(\bar{r}_x)_{\rm tire\ B} = 32,500$  lb.

					Ave	rage val	was						Tire /	1			Tire B		
Rum	D/sq in.	δ <sub>ο</sub> , in.	ē, in.	₩, deg	<b>ř</b> y, lb	1b	F <sub>t</sub> , 1b	∆P <sub>2</sub> , lb-in.	й <sub>х</sub> , lb-in.	ā, in.	Ly, in.	lb/sq in.	δ <sub>0</sub> , in.	о, Н	r <sub>e</sub> , in.	p, lb/sq in.	δ <sub>o</sub> , in.	8, in.	r <sub>e</sub> , in.
72 73 74 75 76	130 130 130 131 135	3.7 3.7 3.5 5.5 5.6	5.4 5.7 5.6 5.9	2.1 2.1 3.9 7.4 10.9	4,650 4,520 8,410 14,380 18,360	500 300 500 2,300 3,800	4,670 4,730 8,420 14,560 18,750	27,100 24,100 56,200 59,500 59,500	50,600 30,400 62,500 65,500 59,700	6.55 6.71 7.42 4.50 3.18	* *   * *	130 130 131 131 135	3.8 3.7 3.7 3.4 3.5	3.4 3.5 3.7 3.4 3.9	26.3 26.3 26.2 26.3	150 151 130 151 135	3.7 3.7 3.5 3.6	3.4 3.6 3.7 3.7 3.9	26.3 26.2 26.2 26.4
77 78 79 80 81 82 83	155 147 148 147 155 144 147	3.1 3.2 3.8  3.1 3.2 3.3	5.1 5.2 5.6 5.6 5.6	.35 .35 .35 .35 14.4 17.9	1,190 1,140 1,250 1,080 20,810 21,760 20,870	* * * 5,000 6,200 6,000	1,190 1,140 1,250 1,080 21,400 22,610 21,700	0 0 0 35,200 32,000 30,600	5,300 5,400 5,400 5,400 41,500 38,400 37,000	1.45 4.74 4.39 1.94 1.70 1.71	** ** ** **	155 148 148 149 155 144 148	3.1 3.1 3.1 3.1 3.2 3.3	3.1 3.1 3.6 3.5 3.5	26.6 26.5 26.5 26.7 27.2	155 146 148 145 153 144 146	3.1 3.3 3.3 3.1 3.3 3.3	5.1 5.3 3.3 3.5 3.6 3.6	26.4 26.4 26.4 26.4 26.4
84 85 86 87 88 89 90	174 173 174 174 173 178 175	2.9 3.1 2.9 2.9 2.9 2.9	2.7 2.9 3.0 3.0 3.1 3.1	2.1 3.9 3.9 7.4 10.9 14.4 17.9	6,070 9,770 10,390 15,690 19,050 20,150 21,160	600 700 700 2,100 3,500 4,800 5,900	6,090 9,800 10,410 15,830 19,370 20,710 21,950	25,100 42,500 38,800 48,200 43,000 52,700 28,700	29,600 47,600 44,100 55,500 48,300 58,400 54,400	4.86 4.86 4.24 5.38 2.49 1.85 1.57	17.8 ** ** ** **	174 175 174 174 173 178 178	3.0 3.2 3.0 2.8 2.9 2.8 3.0	2.6 2.9 3.1 3.1 3.0 3.1 2.9	26.9 26.8 27.0 26.9 26.9 26.8	174 175 174 174 173 178 175	2.8 3.1 2.9 2.9 2.9 2.9 2.9	2.7 2.9 2.9 3.0 3.0 3.1 3.3	26.9 26.6 26.7 26.6 26.6 26.4
91 93 94 95 96 101 102 103 104	232 232 232 232 237 232 233 233 231 232 231 230	255 545544545454	2.255 2.4 2.4 2.4 2.5 5.5 5.5 5.5 5.5 5.5 5.5 5.5 5.5 5.5	.35 .35 .35 .35 .35 2.1 2.1 2.1 3.9 3.9 14.4 17.9	1,20 1,30 1,30 1,30 1,30 1,30 1,30 1,30 1,3	* * * 500 400 700 600 2,000 3,100 4,500 5,300	1,220 1,320 1,270 1,170 1,500 1,040 5,850 10,180 10,910 16,470 19,270 20,350 20,250	00000000000000000000000000000000000000	3,700 4,000 4,000 4,000 18,300 16,200 46,100 56,600 45,300 35,600 36,400 31,900	3.05 3.15 3.06 4.13 3.15 2.79 4.40 3.37 1.85 1.79	** ** ** ** ** ** ** ** ** **	236 238 238 236 237 237 237 230 230 231 230	24.5 - 5.5.5.4 + 5.4.5.4	2.4 2.5 2.5 2.5 2.5 2.5 2.5 2.5 2.5 2.5 2.5	27.2 27.1 27.3 26.8 27.2 27.2 27.2 27.2 27.2 27.2 28.6	232 232 232 232 233 233 231 232 231 232 231 232 231 232	2.25.5.4.5.4.4.5.5.5.5.5.5.5.5.5.5.5.5.5	2.55 - 35555544544	87.2 27.3 27.2 27.2 27.3 27.3 27.3 27.3 2

<sup>\*</sup>Force too small to be measured accurately.
\*\*\*Relaxation length not determined.

TABLE II. - YAW TEST DATA - Continued (d) SERIES D:  $\bar{F}_z$  = 39,800 lb;  $(F_z)_{tire\ A}$  = 39,900 lb;  $(F_z)_{tire\ B}$  = 39,700 lb.

					Ave	rage va	luos						Tire	A			Tire	В	
Rum	ī, lb/sq in.	ā,, in.	ð, in.	ų, deg	Fy, 1b	Ē <sub>x</sub> , lb	Γ <sub>ψ</sub> , 1b	ΔM̄ <sub>z</sub> , lb-in.	Йg, lb-in.	đ, in.	Îy, in.	P, lb/sq in.	ō <sub>o</sub> , in.	ð; in.	r <sub>e</sub> ,	p, lb/sq in,	δ <sub>o</sub> , in.	ð, in.	re,
105 106 107 108 109 110	128 130 127 128 126 126	4.0 4.1 4.2 4.2	5.8 5.8 4.1 4.1 4.5	2.1 2.1 3.9 3.9 7.4 10.9	3,780 3,580 6,980 6,800 13,510 17,910	600 700 800 1,200 2,900 5,100	3,800 3,600 7,020 6,870 13,770 18,550	38,700 38,100 70,200 65,100 78,000 67,000	45,100 44,600 77,500 72,200 85,500 75,000	11.87 12.39 11.04 10.51 6.21 4.04	** : : : * *	128 131 127 128 126 126	4.0 4.0 4.3 4.3 4.0	5.6 5.9 4.3 4.5 4.5	26.1 26.1 26.1 26.0 26.1 26.1	128 129 127 128 126 127	4.1.4.4.4.4.4.4.4.4.4.4.4.4.4.4.4.4.4.4	4.07 4.29 4.5 4.5	26.0 26.0 25.9 25.9 26.0 26.1
112 113 114 115	151 153 150 150 152	3.5 3.5 3.7 3.7 3.6	3.3 3.5 3.5 4.1	2.1 2.1 3.9 7.4 10.9	4,790 4,810 8,510 15,350 20,220	600 600 1,100 2,900 4,900	4,810 4,830 8,370 15,600 20,780	32,100 34,200 60,600 70,800 63,800	37,800 39,900 66,800 77,800 71,600	7.86 8.26 7.98 4.99 3.45	**	151 153 150 150 152	3.5 3.5 3.7 3.8 3.6	3.4 3.4 3.6 3.8 4.2	26.4 26.4 26.2 26.5 26.5	150 153 150 150 151	3.5 3.7 3.7 3.7	3.2 3.5 3.9 4.0	26.4 26.4 26.2 26.4 26.5
116 117 118 119 120 121 123 124 125 127 128 129	232 230 232 230 231 230 231 231 231 231 231 231 231 231	2.77776677687777777777777777777777777777	2.77.76.66.66.66.788.7.28.28.28.28.28.28.28.28.28.28.28.28.28.	.35. .35. .35. .35. .35. .35. .35. .35.	1,190 1,320 1,220 1,230 1,280 6,830 6,790 11,100 10,990 18,570 22,620 22,810 21,930	* * * * * * * * * * * * * * * * * * *	1,190 1,320 1,220 1,210 1,220 1,280 6,830 11,120 11,010 18,780 22,530 23,680 23,680	24,400 24,400 52,300 50,400 50,400 43,500 24,700	5,300 5,300 5,300 5,300 5,000 5,000 29,400 57,300 55,100 55,700 49,600 32,200 30,000	4.45 4.58 4.10 4.57 4.15 6.27 4.55 6.29 6.29 6.29 6.29 6.29 6.29	***	252 259 259 250 251 251 251 251 252 252 253 253 253 253 253 253 253 253	2.7862.6675.967687	2.8676664566787	27.2 27.1 27.2 27.1 27.2 27.0 27.1 27.2 26.9 27.2 27.2 27.3 27.3 28.2	231 230 230 231 231 231 231 231 231 231	2.6 2.7 2.8 2.6 2.7 2.8 2.7 2.8 2.7 2.8 2.7 2.8 2.7 2.8	2.677866658669987	27.1 27.0 27.0 27.0 27.0 27.0 27.0 27.0 27.0

<sup>\*</sup>Force too small to be measured accurately.
\*\*\*Relaxation length not determined.

TABLE II .- YAW TEST DATA - Continued (e) SERIES E:  $\bar{r}_z = 45,200 \text{ lb}$ ;  $(r_z)_{\text{tire A}} = 45,400 \text{ lb}$ ;  $(r_z)_{\text{tire B}} = 45,000 \text{ lb}$ .

					Avex	age valu	<b>10</b> 8						Tire /	1			Tire :	В	
Run	p, lb/sq in.	Б <sub>о</sub> , in.	ē, in.	₩, dag	<b>f</b> y, 1b	ř <sub>x</sub> , 1b	P <sub>ψ</sub> , 1b	Ƅ <sub>z</sub> , 1b-in.	H, lb-in.	ō) ii.	L <sub>y</sub> , in.	p, lb/sq in.	δο, in.	8, in.	r <sub>e</sub> , in.	p, 15/8q in.	δ <sub>0</sub> , in.	ð, in.	re, in.
150 151 132 135 134 135	151 152 152 151 150 150	4.0 4.1 4.1 4.2	3.8 3.8 4.0 3.9 4.7	2.1 2.1 3.9 5.9 7.4 10.9	4,900 4,730 8,460 8,480 14,860 19,990	900 800 1,000 1,400 3,100 4,900	4,950 4,760 8,510 8,500 15,140 20,560	40,400 40,900 74,700 74,400 86,300 84,100	47,400 47,900 82,200 81,700 93,700 93,400	9.61 10.66 9.66 9.61 6.19	## 20.0	151 150 153 151 150 152	5.59 4.0 4.0 4.0	7770895	26.1 26.2 26.1 26.3 26.3	151 154 150 150 150 150	4.0 4.1	5,5,4,0 4,4,4,4,4,4,4,4,4,4,4,4,4,4,4,4,4,4	26.1 26.1 26.2 26.2
136 137 138 139 140 141 142 143	183 180 183 181 182 181 180 180	5.6 3.7 5.6 5.6 5.5 5.5 5.5	5.6 5.9 5.4 5.4 5.7 6	2.1 2.1 3.9 5.9 7.4 10.9	1,090 5,510 5,830 9,880 9,850 9,510 17,280 28,290	* 800 700 600 1,000 1,000 3,700 5,400	1,090 5,540 5,850 9,900 9,900 9,560 17,610 22,910	0 39,300 41,100 57,600 60,900 49,900 79,800 65,600	7,200 46,100 47,100 64,200 67,500 56,500 87,200 72,700	6.61 8.32 8.05 6.48 6.82 5.91 4.95	## ## ##	183 180 183 181 180 181 180 180	5.7466465 5.75555555555555555555555555555555555	65144588 555555555	86.7 86.6 86.6 86.6 86.6 86.7	185 180 185 180 184 181 179 180	5.6 3.7 3.4 5.5 3.5 3.4	5.6 3.5 5.2 5.3 5.7 5.7	26.6 26.5 26.4 26.5 26.5 25.9
144 147 148 149 150 153 155 156 157 158 160 161 162	255 252 252 252 252 252 252 252 253 252 252	250,909,990809289990919	2.9 3.9 3.9 3.9 2.9 3.8 3.8 3.8 3.8 3.8 3.9 3.9 3.9 3.9 3.9 3.9 3.9 3.9	.35 .35 .35 .35 .35 .35 .35 .35 .35 .31 2.1 3.9 7.4 10.9 10.9 10.9 14.4	1,260 1,310 1,350 1,210 1,410 1,210 1,120 1,520 6,160 11,350 11,540 19,860 19,870 25,340 22,710 24,660 24,660	* * * * * * 600 \$,000 1,000 1,200 5,300 5,000 5,600 6,600 8,300	1,860 1,310 1,350 1,410 1,410 1,220 1,320 6,510 6,170 11,390 11,590 20,110 19,850 23,860 23,860 23,850 25,650	0 0 0 0 0 0 0 32,000 28,700 49,200 48,200 48,200 65,300 61,800 57,900 57,900 57,900	6,000 6,300 6,000 6,000 6,000 6,000 6,000 57,700 34,400 71,500 71,000 68,100 60,200 59,400 39,300	4.76 4.44 5.46 5.54 5.75 4.77 5.78 7.75 2.55 2.55 1.75	**  **  **  **  **  **  **  **  **  **	252 231 232 231 232 232 232 233 232 232 23	908099990808588999919	9080999085991799089	27.0 27.0 27.0 27.0 27.0 27.0 27.1 27.1 26.8 27.0 27.2 27.2	255 232 252 252 252 252 253 250 252 254 252 252 252 252 252 252 252	25.90909085.91	2.91.90.90 2.0.90.88 2.71.90.11 2.50.11 3.0.88	27.0 26.9 26.9 27.0 27.0 27.0 27.0 27.0 27.0 27.0

<sup>\*</sup>Force too small to be measured accurately.
\*\*Relaxation length not determined.

TABLE II. - YAW TEST DATA -- Continued

(f) SERIES F:  $\tilde{F}_{x} = 17,400 \text{ lb; } (F_{x})_{\text{tire A}} = 17,500 \text{ lb; } (F_{x})_{\text{tire B}} = 17,400 \text{ lb.}$ 

D					Ave	rage val	ules						Tire	A			Tire	В	
Run	ē, lb/sq in.	ō, in.	5, in.	₩, deg	fy, lb	Ē <sub>x</sub> , lb	Γţ,	ΔΩ, lb-in.	R, lb-in.	ţ, in.	ī, in.	lb/sq in.	å,,	ð, in.	r <sub>e</sub> ,	p, lb/sq in.	δ <sub>o</sub> , in.	8, in.	r <sub>e</sub> ,
163 164 165 166 167 168 169	61 60 59 60 59 60 60	5.4 5.6 5.4 5.5 5.4 5.4	3.4 3.3 3.5 3.6 3.9 4.1	0.35 2.1 3.9 7.4 10.9 14.4 17.9	550 2,680 4,690 8,760 11,070 12,850 13,060	* 300 800 1,700 2,700 4,000 4,800	530 2,690 4,730 8,910 11,380 15,440 13,900	0 18,000 32,900 34,300 32,700 29,000 25,500	4,100 22,000 37,200 38,800 37,700 34,400 29,200	7.74 8.18 7.86 4.35 3.31 2.56 2.10	**	61 60 59 59 59 59	5.3 5.5 3.6 5.3 3.5 5.4	3.5 3.6 3.8 3.8 4.5	25.8 25.9 25.7 26.1 25.9 25.9 26.2	62 60 59 60 59 60 60	5.4 3.5 3.6 3.5 5.4 3.4	3.4 3.5 3.4 3.4 4.0 4.1 4.2	25.8 25.8 25.8 25.9 25.8 25.9 25.4
170 171 172 173 174 175	ವರ <b>್</b> ವಿಪರಿ	2.6 2.7 2.6 2.7 2.8 2.8	2.5 2.7 2.8 3.0 3.1 3.2	2.1 3.9 7.4 10.9 14.4 17.9	3,470 6,070 9,980 11,740 13,010 13,310	300 600 1,700 2,800 3,300 4,200	3,480 6,100 10,120 12,060 13,420 13,960	17,100 29,500 28,000 24,100 22,300 18,400	19,900 52,700 31,500 27,900 26,300 26,600	5.72 5.36 5.11 2.31 1.96 1.62	**	84 85 84 84 84	2.6 2.7 2.4	2.6 2.7 2.5 3.1 3.1 3.2	26.5 26.4 26.6 26.9	84 84 85 84 85 84	2.6 2.8 2.7 2.7 2.8 2.8	2.4 2.7 3.0 2.8 3.1 3.2	26.4 26.3 26.4 26.2 26.2 25.8
176 177 178 179 180 181 182 183 184 185 186 187 188 188 189	170 167 169 168 169 169 168 169 168 168 168	1.7 1.8 1.7 1.8 1.7 1.7 1.7 1.7 1.7 1.7	1.7 1.8 1.7 1.8 1.7 1.6 1.5 1.7 1.7 1.7	.35 .35 .35 .35 .35 .35 .3.9 .3.9 .3.9	830 760 880 740 850 850 4,050 6,150 10,150 11,520 12,120 12,120	* * * * * 100 500 500 1,500 2,500 3,100 3,700	850 760 880 740 810 850 4,050 6,850 7,370 10,560 11,750 12,720	0 0 0 0 0 0 8,700 12,200 16,700 20,800 19,200 11,300 11,300 10,100	2,000 2,200 2,200 2,200 2,000 10,500 14,400 18,500 22,400 20,800 17,200 13,300 12,500	2.41 2.89 2.97 2.72 2.75 2.75 2.55 2.51 1.97 1.06	** ** ** ** ** ** ** ** ** **	170 167 169 169 169 169 169 169 169 169	1.6 1.9 1.8 1.7 1.9 1.7 1.7 1.8	1.9689 5.9545657	27.5 27.5 27.5 27.4 27.4 27.4 27.4 27.4 27.4 27.4	170 166 169 168 169 169 167 169 168 168 169	1.7 1.8 1.8 1.7 1.7 1.6 1.6 1.7	1.7 1.7 1.8 1.8 1.7 1.6 1.8 1.7 1.7	27.4 27.4 27.5 27.5 27.5 27.5 27.5 27.5 27.5 27.5

<sup>\*</sup>Force too small to be measured accurately. \*\*\*Relaxation length not determined.

TABLE II.- YAW TEST DATA - Continued (g) SERIES G:  $\vec{F}_x = 9,600$  lb;  $(F_x)_{tire\ A} = 9,700$  lb;  $(F_x)_{tire\ B} = 9,600$  lb.

					Avar	aga valu	101						Tire !				Tire l	В	
Run	D/sq in.	ā, in.	B, in.	₩, deg	ё <sub>у</sub> , 1ь	ř <sub>х</sub> , њ	Р <sub>ψ</sub> , 1ь	Ƅ <sub>2</sub> , lb-in.	A <sub>z</sub> , 16-in.	ā, in.	L <sub>y</sub> , in.	lb/sq in.	å,, in.	ð, in.	re, in.	lb/sq in.	ð <sub>o</sub> , in.	ð, in.	r <sub>e</sub> , in.
190 191 192 193 194 195 196	ଅନ୍ଧନ୍ଧ ଅନ୍ଧନ୍ଧ ଅନ୍ଧନ୍ଧ ଅନ୍ଧନ୍ଧ ଅନ୍ଧନ୍ଧ ଅନ୍ଧନ୍ଧ ଅନ୍ଧନ୍ଧ ଅନ୍ଧନ୍ତ ଅନ୍ଧନ୍ଧ ଅନ୍ଧନ୍ତ ଅନ୍ଧନ୍ତ ଅନ୍ଧନ୍ତ ଅନ୍ଧନ୍ତ ଆଧାର ଅନ୍ଧନ୍ତ ଆଧାର ଆଧାର ଆଧାର ଆଧାର ଆଧାର ଆଧାର ଆଧାର ଆଧାର	3.6 5.4 3.5 3.5 3.5 3.4	3.6 3.2 3.6 3.7 3.9 3.9	0.35 2.1 3.9 7.4 10.9 14.4 17.9	450 1,710 3,100 6,310 7,590 8,690 8,560	* 200 300 1,000 1,400 2,100 2,600	1,720 3,110 6,390 7,720 8,940 8,940	0 6,500 20,100 25,700 21,700 21,700 6,900 7,400	3,88 9,780 23,580 23,580 23,780 23,780 11,780	8.44 5.64 7.56 4.62 3.33 1.25 1.31	**	ଞ୍ଚଳନ୍ଦନ୍ଦନ୍ଦନ୍ଦ	55555555	5.6 5.1 5.7 5.7 5.0 5.9	25.6 25.7 25.5 25.7 25.5 26.0 26.2	89 89 89 89 89 89 89 89	3.74.66 5.50 5.50 5.50 5.50 5.50 5.50 5.50 5	3.7 3.3 3.5 3.5 3.7 4.0	25.5 25.5 25.4 25.7 25.7 25.7 25.7
197 198 199 200 201 202 203	39 39 39 39 39 39 39	2.606	2.6 2.5 2.9 2.8 3.3 3.0 3.2	2.1 3.9 7.4 10.9 14.4 17.9	480 1,860 3,640 6,670 8,100 8,860 8,530	* 200 400 900 1,400 1,900 2,500	1,870 3,660 6,730 8,220 9,050 8,890	0 8,500 17,500 19,100 18,400 9,400 8,600	2,400 10,800 20,400 21,900 22,000 12,500 12,000	5.00 5.78 5.57 5.25 2.68 1.38 1.35	**	39 39 39 39 39 39 39	2.6 2.6 3.1 2.6 2.5 2.8	2.6 2.3 3.0 2.7 3.0 3.0 3.0 3.0	26.0 26.9 25.9 25.9 26.7	59 59 59 59 59 59 59	2.7 2.7 3.0 2.7 2.6 2.8	2.7 2.6 2.9 2.8 5.5 3.2	25.9 26.0 25.9 25.9 25.9 25.5
204 205 206 207 208	59 59 59 59 59	2.0 1.9 2.4 1.9 2.1	2.0 1.8 2.2 2.1 2.3	2.1 2.1 3.9 7.4 10.9	630 2,480 4,690 7,270 8,110	200 400 1,000 1,500	630 2,490 4,710 7,340 8,250	0 5,500 14,600 16,000 15,100	1,700 6,700 16,600 17,800 17,300	2.70 2.69 3.52 2.43 2.10	## ## ## ##	59 59 59 59 59	2.0 1.9 2.5 1.8	2.0 1.8 2.1 2.3 2.4	26.3 26.6 26.3 26.7 26.6	58 59 59 59 59	2.1 1.9 2.3 1.9 2.1	2.1 1.7 2.2 1.8 2.1	26.4 26.7 26.4 26.6 26.3
517 517 517 510 509 509	84 84 84 84	1.7 1.6 1.6 1.6 1.7 1.6	1.7 1.6 1.6 1.4 1.7	.35 .35 .35 2.1 3.9 7.4	630 640 640 2,700 4,890 7,540	* * 200 300 800	630 640 640 2,710 4,900 7,580	0 0 0 14,500 15,700	1,400 1,500 1,500 15,700 14,800	2.22 2.05 2.05  3.20 1.95	**	84 84 84 84 84 84	1.7 1.6 1.6 1.8 1.7	1.7 1.7 1.6 1.5 1.6	26.7 26.9 27.1 26.7 27.0	84 84 85 85 84 84	1.6 1.6 1.7 1.6 1.6	1.6 1.6 1.4 1.7 1.6	26.8 26.9 27.0 26.8 27.0
215 216 217 218	127 126 126 126	1.3 1.2 1.5 1.3	1.3 1.2 1.4 1.3	.55 2.1 3.9 7.4	2,460 2,460 4,910 6,700	* 200 300 700	550 2,470 4,920 6,730	16,100 11,800	1,100  17,300 12,900	2.00 3.52 1.92	**	127 127 126 126	1.3 1.2 1.5 1.4	1.3 1,1 1.3 1.4	27.0 27.5 27.1 27.2	127 125 126 126	1.5 1.3 1.5 1.3	1.3 1.3 1.5 1.2	27.2 27.3 27.2 27.4
21.9 22.1 22.2 22.2 22.2	168 168 166 168	1.0 1.0 1.2 1.1	1.0 1.0 1.2 1.0	.35 2.1 3.9 7.4	500 2,440 4,250 6,310	* 100 300 600	500 2,1440 4,260 6,330	9,700 11,000	10,800 11,800	1.60 2.54 1.86	**	168 168 166 167	1.0 1.0 1.3	1.0 .9 1.5	27.5 27.6 27.3 27.4	168 168 165 168	1.0 1.0 1.2 1.1	1.0 1.0 1.0 1.0	27.0 27.3 27.5 27.6

<sup>\*</sup>Force too small to be measured accurately.

<sup>\*\*</sup>Relaxation length not determined.

TABLE II.- YAW TEST DATA - Continued (h) SERIES H:  $F_z$  = 9,600 lb;  $(F_z)_{tire\ A}$  = 9,700 lb;  $(F_z)_{tire\ B}$  = 9,600 lb.

					Average	values				
Rum	p, lb/sq in.	δ <sub>ο</sub> , in.	δ, in.	₩, deg	Ēy, lb	F <sub>x</sub> , 1b	F <sub>\psi</sub> , 1b	ΔΜ <sub>z</sub> , lb-in.	$\bar{M}_z$ , lb-in.	ā, in.
224 224 225 227 237 237 237 237 237 237 237 237 237	**************************************	44646 5 566 56 5 7999999 78 1910 00 00 00 00 00 00 00 00 00 00 00 00 0	446 544 556 556 79999956 0790 1930 1	0.35 0.35 2.1 2.19 2.19 2.19 2.19 2.19 2.19 2.11 2.35 2.11 2.35 2.11 2.35 2.11 2.35 2.11 2.35 2.11 2	480 450 450 1,660 1,550 3,960 3,960 3,960 3,870 4,870 4,870 5,870 5,800 5,980 5,980 5,980 2,1560 2,2660 2,3,220 2,3,220 2,3,220 2,3,260 2,100	* * * 200 200 300 400 500 800 1,200 1,600 * * * * * * * 100 300 200 400 600 1,600 1,800 1,800	480 450 450 1,670 1,560 2,740 3,980 2,740 5,140 5,140 5,140 5,140 5,140 2,120	0 0 0 11,300 12,800 9,800 13,000 17,900 14,000 13,800 12,100 9,700 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	3,500 3,500 3,800 14,700 16,300 13,300 16,400 21,600 17,500 15,800 15,800 2,900 2,900 2,900 2,900 2,900 2,900 2,900 1,500 1,500 12,500 12,500 12,500 12,500 12,900	7.744 8.455 8.455 8.477 8.477 8.477 8.477 8.477 8.477 8.477 8.477 8.477 8.477 8.557 8.477 8.557 8.477 8.557 8.477 8.557 8.557 8.477 8.557 8.477 8.557 8.477 8.557 8.477 8.557
251 252	39 39	2.9 3.0	3.1 3.2	14.4 17.9	5,700 4,550	1,700 1,900	5,940 4,910	7,200 7,400	10,400 10,800	1.75 2.20

<sup>\*</sup>Force too small to be measured accurately.

TABLE II. - YAW TEST DATA - Concluded

(i) SERIES I:  $\overline{F}_z = 9,600$  lb.

Run		Average values												
	p, lb/sq in.	$\bar{\delta}_0$ , in.	ē, in.	ų, deg	Ēy, 1b	ξ <sub>x</sub> , lb	F̃ <sub>ψ</sub> , lb	$\Delta ar{M}_{Z}$ , lb-in.	M <sub>z</sub> , lb-in.	ā, in.				
253 254 255 256 257 258 259 260 261 262 263 264 265 266 267	39 39 39 39 39 39 39 39 39	2.9 2.8 2.8 3.8 3.0 3.0 3.0 3.0 3.0 3.0 3.0 3.0 3.0 3.0	9-8876183042332 2-2-2-3-2-3-3-3-3-3-3-3-3-3-3-3-3-3	0.35 .35 .35 2.1 2.9 3.9 7.4 10.9 14.4 17.9	570 510 620 590 2,490 4,440 4,170 7,120 6,710 8,220 7,970 8,380 8,260 8,130	*  *  *  300  400  500  600  1,100  1,200  1,800  1,800  2,400  2,500  2,600	570 510 620 590 2,500 4,460 4,200 7,200 6,810 8,410 8,170 8,710 8,540	0 0 0 17,000 16,300 19,100 16,900 17,500 16,900 11,000 13,200 7,500 6,500 4,700	2,900 2,800 2,800 19,600 18,700 22,300 19,700 21,100 20,000 14,700 16,600 11,100 8,100	5.09 4.52 4.75 7.48 5.00 4.69 2.94 1.75 2.03 1.27 1.17 .95				

<sup>\*</sup>Force too small to be measured accurately.

TABLE III. - STATIC-RELAXATION-LENGTH DATA

			· Ti	re A				Tire B					
Run	Test series	p <sub>o</sub> , lb/sq in.	p, lb/sq in.	F <sub>z</sub> , lb	r, in.	δ <sub>o</sub> , in.	L <sub>s</sub>	p <sub>o</sub> , lb/sq in.	p, lb/sq in.	F <sub>z</sub> ,	r, in.	8₀ in.	L <sub>g</sub>
268 269 270 271	A A A	28 56  146	29 58  148	17,000 17,000 17,000 17,000	27.5 27.6  27.8	5.1 5.8  2.2	20.9 26.5 26.9	29 59 95 146	31 63  148	16,900 16,900 16,900 16,900	27.5 27.7 27.8 27.9	5.2 3.7 2.7 2.2	20.5 25.3 31.1 28.2
272 273 274	В В В	78 122 202	79 124 203	23,700 23,700 23,700	27.6 27.7 27.9	3.8 2.9 2.1	25.3 26.1 22.5	78 122 202	79 124 203	23,800 23,800 23,800	27.7 27.8 27.9	3.8 2.9 2.0	19.8 24.3 25.6
275 276 277	G G	· 128 171 229	173	32,900 32,900 32,900	27.9 28.0 28.1	3.5 2.9 2.5	20.3 23.0 28.0	128 171 230	173	32,500 32,500 32,500	28.0 28.1 28.2	3.7 3.1 2.5	22.1 23.7 20.7
278 279 280	ם ס ס	126 147 231		39,900 39,900 39,900	27.9 28.0 28.1	4.1 3.8 2.7	18.7 22.8 20.9	125 148 231		39,700 39,700 39,700	27.9 28.0 28.2	4.1 3.8 2.8	15.8 22.2 19.8
281 282 283	E B B	143 177 228	148 184 231	45,400 45,400 45,400	28.0 28.1 28.2	4.5 3.7 3.3	18.9 20.2 22.5	143 177 228	148 183 234	45,000 45,000 45,000	28.0 28.1 28.2	4.3 3.8 3.3	18.9 19.3 19.0
284 285	F F	57 162	59 164	17,300 17,300	27.7 27.9	3.6 1.9	20.8 24.5	58 161	60 162	17,400 17,400	27.7 27.9	3.6 1.9	20.6 26.7
286 287	H H	29 38	29 39	9,700 9,700	27.2 27.2	3.5 2.8	25.6 29.7	28 36	29 37	9,600 9,600	27.3 27.3	3.3 2.8	23.6 23.7

TABLE IV .- UNYAWED-ROLLING RELAXATION-LENGTH DATA

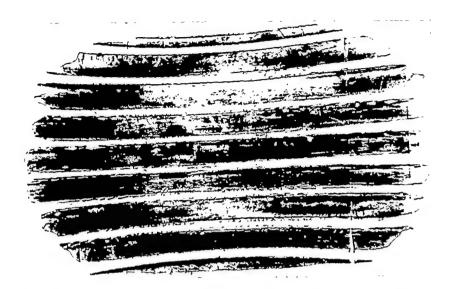
Run		Average values							
	Test series	p, lb/sq in.	F <sub>z</sub> , lb	ō, in.	ī <sub>f</sub> , in.	$f_{\lambda}$ , in.			
288	A	58 60	17,000	3.5	31.6	23.8			
289	A	60	17,000	3.7	27.0				
290 [	A.	60	17,000	3.7	23.3	24.8			
291	Ā	62 62	17,000	3.7	23.3 24.4	20.3			
292	A	62	17,000	3.7	23.3				
293	Ā	104	17,000	3.7 2.4					
294	A	150	17,000	2.0					
291 292 293 294 295	Ā	150	17,000	2.1					
296	В	79	23,700	3.8	, ====				
297	В	79 126	23,700	2.7	16.4	21.3			
296 297 298	B B B	203	23,700	2.3	20.4				
299	C	130	32,700	3.7	25.2				
300	C	131	32,700	3.7	22.9	l			
301	0 0 0	173	32,700	3.1	23.4				
302	Ċ	172	32,700	3.1	21.9				
303	C	231	32,700	2.6	25.3				
303 304	C	229	32,700	2.6	21.0				
305 306	D	126	39,800	4.2	14.0				
306	D	126	39,800	4.2	19.5				
307	D	150	39,800 39,800	3.6	19.7				
307 308	D	150	39,800	3.7 2.8	19.5				
309	D	232	39,800	2.8	22.6				
309 310	ם מ ם ם ם ם	232	39,800	2.8	28.3				
311	ם	150 232 232 232	39,800	2.8	24.8				
312	E	150	45,200	,4.2	18.8				
313	E	150	45,200	4.2	18.8				
314	E	1.80	45,200	3.7	18.3				
313 314 315	E	1.80	45,200	3.8	17.1				
316	e e e e	232 232	45,200	3.1	23.2 18.5				
317	Z	` 232	45,200	3.1	18.5				

TABLE V.- LOCKED-WHEEL DRAG TEST DATA  $\begin{bmatrix} \bar{F}_{2} = 9,730 \text{ lb for } \bar{F}_{X} = 0 \end{bmatrix}$ 

		verage	values.		(-)	(2)	Tire A		Tire B			
Rum	ρ, lb/eq in.	Eo,	μ <sub>x</sub>	Ē <sub>x</sub> , lb/in.	(F <sub>E</sub> ) <sub>tire A</sub> + (F <sub>E</sub> ) <sub>tire B</sub> ,  1b	(F <sub>X</sub> ) <sub>tire A</sub> + (F <sub>X</sub> ) <sub>tire B</sub> ,	lb/#q in.	ôo, in.	lb/sq in.	ōo, in.	Remarks	
318 319	19	4.6	0.81	7,800 7,160	18,450 18,410	14,870 15,460	19 19	4.6 4.7	19 19	4.6 5.0	•	
520 521 522 323 524 325	89 89 89 89 89 89 89	3.6 3.4 3.7 5.7	.88 .82 .88 .88 .74	8,640 7,700 6,750 7,700 6,900 8,100	18,360 18,430 18,360 18,360 18,530 18,540	16,190 15,070 16,200 16,220 13,700 13,500	29 29 30 29 29 29	5.8 5.4 5.6 5.6 5.6	29 29 30 29 29 28	\$.0 3.7 5.5 3.8 3.6 3.8	Wet concrete Wet concrete	
326 327	39 39	2.8	.81	8,300 8,130	18,440 18,340	14,950 16,510	39 39	2.8 3.3	39 39	2.7 3.4		
328 329	59 59	2.6	· 79 .86	8,900 9,590	18,470 18,390	14,590 15,730	59 59	2.6	59 59	2.6		
330 331 332 333 334 335	84 84 85 84 84 84	1.7 2.1 1.6 1.6 1.6	.80 .85 .74 .73 .84	8,310 8,450 7,600 7,770 6,510 8,340	18,460 18,400 18,550 18,540 18,410 18,350	14,700 15,560 13,680 13,570 15,420 16,250	ಸಿಕು ಕಿನಕ <b>ಿ</b>	1.7 2.1 1.6 1.6 1.6 1.7	84 84 85 84 84 84	1.6 2.0 1.6 1.7 1.7	Het concrete Het concrete	
336 337 338 339	126 126 126 126	1.3 1.3 1.4 1.6	.74 .69 .79 .80	9,400 10,250 8,250 8,300	18,550 18,580 18,470 18,460	15,720 12,900 14,600 14,700	126 126 126 126	1.3 1.4 1.7	126 126 126 126	1.3 1.4 1.6		
340 341	150 150	1.2	.71 .75	9,780 9,100	18,570 18,520	13,160 13,840	150 150	1.2	150 150	1.1		
542 343	168 168	1.1	.72	9,370	18,550 18,540	13,590 13,490	168 168	1.1	168 168	1.1	Ì	
344 345 346 347 348	180 180 180 180 180	1.0 1.3 1.1 1.1	.81 .71 .76 .64	8,030 7,770 7,320 7,260 6,990	18,450 18,560 18,500 18,650 18,590	14,880 15,260 14,140 11,950 12,000	180 180 180 180 180	1.0 1.3 1.1 1.1 1.1	180 180 180 180 180	1.0 1.3 1.1 1.0 1.0	Wet congrete	
349 350	202	1.0	.76	7,450	18,500 18,570	14,070 13,160	505 501	1.1	501 505	1.1		

TABLE VI. - TIRE FOOTPRINT DATA

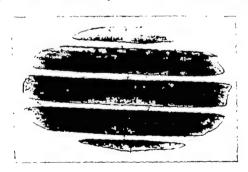
Run	Test series	Po,	p, lb/sq in.	F <sub>z</sub> ,	₿₀,	Ag, in.2	A <sub>n</sub> ,	b, in.	2h,
		lb/sq in.	10/вд 111.	Tire A	ш.	1n	111,-		
351 352 353	C C	152 152 152	152 153 156	32,900	0.7 1.7 3.4	147 106 2140	33 85 173	5.4 8.9 12.9	10.5 16.3 22.3
354 355 356 357 358 359	After I After I After I After I After I After I	199 179 168 150 59	200 180 170 151 59 19	8,620 8,620 8,620 8,620 8,620 8,620	1.0 1.1 1.2 1.3 2.0 3.7	59 64 66 74 127 274	47 50 51 61 94 190	6.7 6.9 7.0 7.8 9.6 13.2	11.3 11.8 11.7 12.4 15.3 23.4
				Tire B					
360 . 361. 362 363 364 365 366	000000	180 180 180 135 135 135 135	180 180 182 135 136 137 138	32,500  32,500	1.1 1.8 2.8 1.0 1.9 2.8 3.8	81 130 207 50 116 186 248	57 84 135 29 75 120 166	7.2 9.2 11.9 5.6 8.7 11.8	13.0 16.5 20.4 10.1 15.5 19.1 22.2
567 568 369 370 372 372 373 375 377 377 377 377 378 380 381	After I	200 200 180 166 150 150 126 83 83 83 57 28 17	200 200 181 167 151 127 84 85 85 58 59 19	8,560 8,560 8,560 8,560 8,560 8,560 8,560 8,560 8,560 8,560 8,560 8,560	1.0 1.2 1.1 1.3 1.3 1.5 1.6 1.7 1.7 1.9 2.6 5.1 4.0	60 58 61 65 69 69 77 104 108 111 139 190 221 304 319	43 44 48 48 57 68 79 137 195 221	6.7 6.8 6.9 7.0 6.9 7.2 8.9 9.1 10.0 11.8 13.7	11.3 11.4 11.5 12.0 11.9 12.1 14.7 14.4 14.9 16.5 16.7 20.6 24.3 24.7



(a) Run 359;  $\delta_0 \approx 3.7$  inches; p = 19 pounds per square inch.

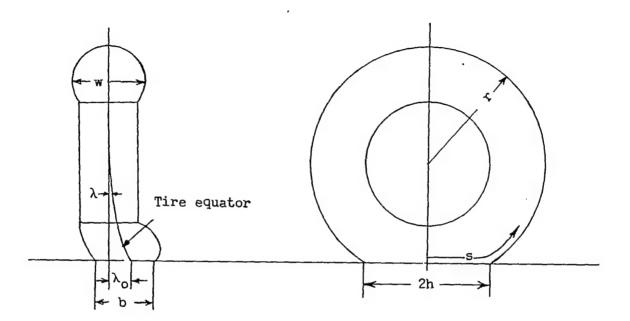


(b) Run 358;  $\delta_0 = 2.0$  inches; p = 59 pounds per square inch.

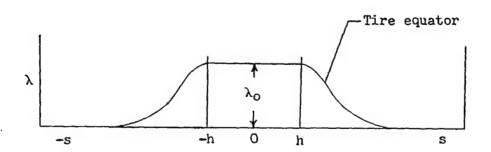


I-84904

(c) Run 356;  $\delta_0$  = 1.2 inches; p = 170 pounds per square inch. Figure 1.- Typical tire footprints for tire A at  $F_z$  = 8,620 pounds.

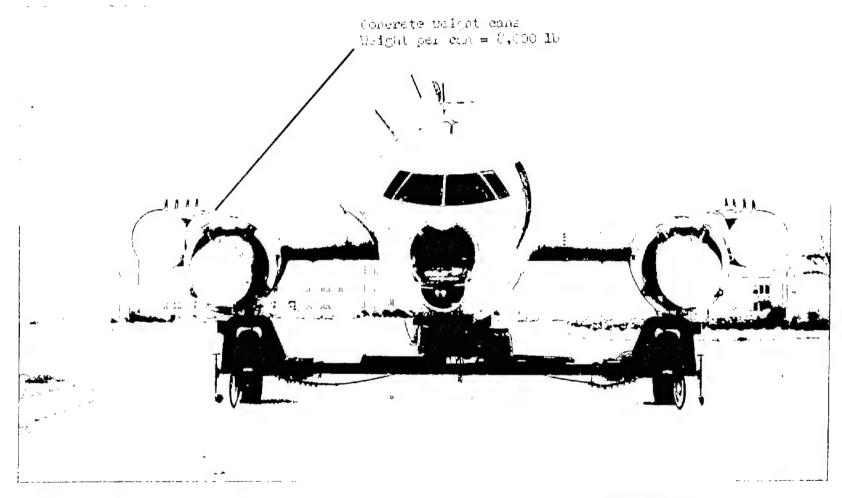


(a) Tire dimensions.



(b) Shape of distortion curve.

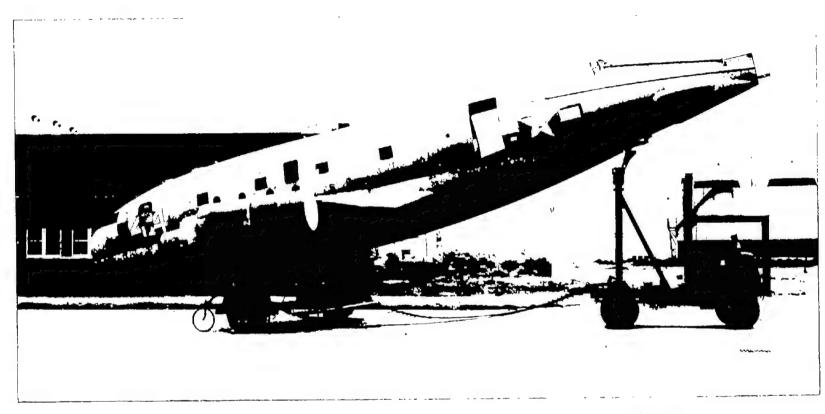
Figure 2.- Tire dimensions and tire distortion.



L<del>-</del>79570.1

(a) End view.

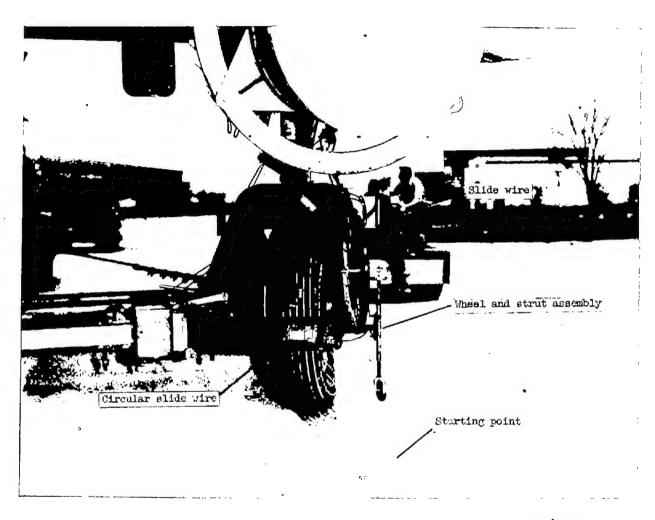
Figure 3.- Test vehicle.



(b) Side view.

Figure 3.- Concluded.

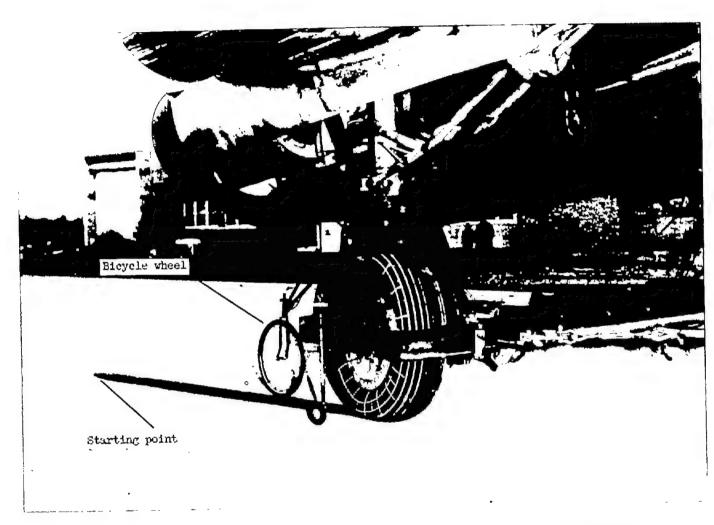
L-79571



1-79491.1

(a) Rear view.

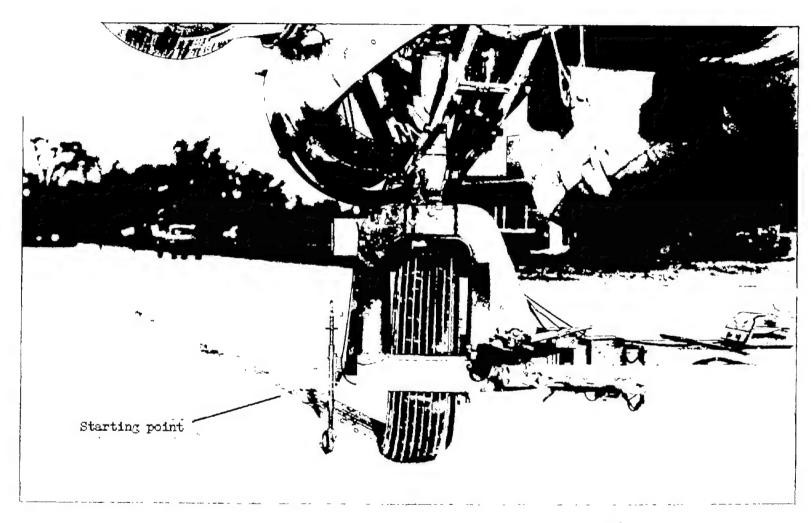
Figure 4.- Tire A under yawed rolling at  $\psi = 17.9^{\circ}$ .



(b) Side view.

Figure 4.- Continued.

L-79492.1



(c) Front view.

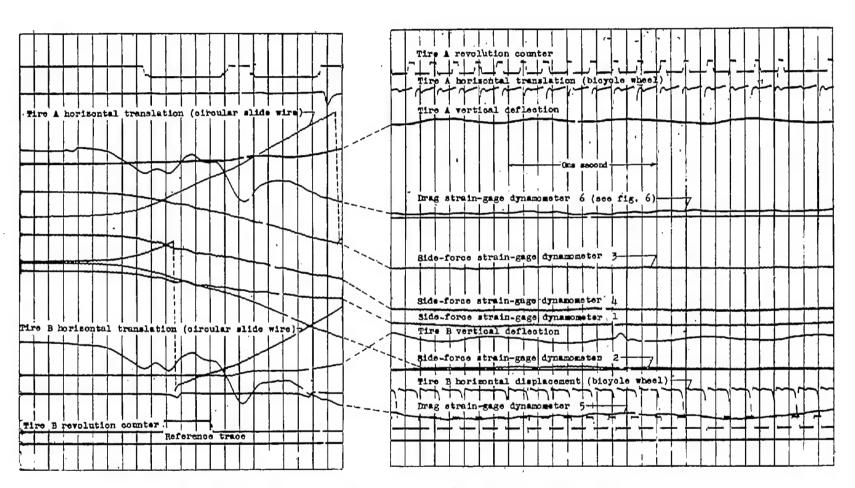
Figure 4.- Concluded.

L-79493.1



Figure 5.- Locked-wheel drag-test setup.

L-80727.1



- (a) Initial-force-buildup stage of run.
- (b) Steady-state stage of run.

Figure 6.- Typical oscillograph record. Run 47; p = 81 pounds per square inch;  $\delta_0$  = 3.9 inches;  $\psi$  = 17.9°.

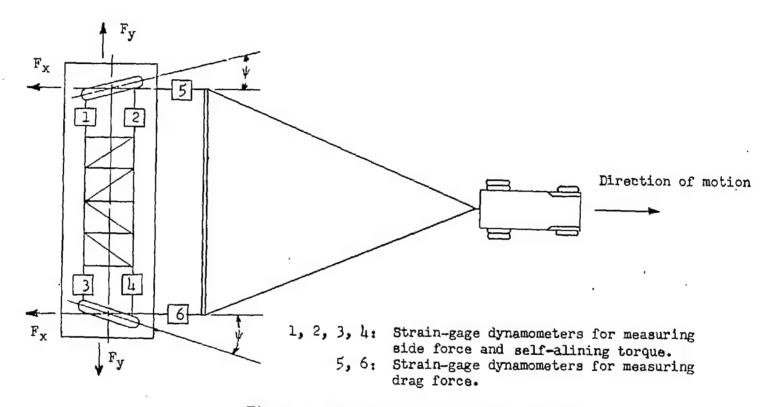


Figure 7.- Strain-gage dynamometer location.

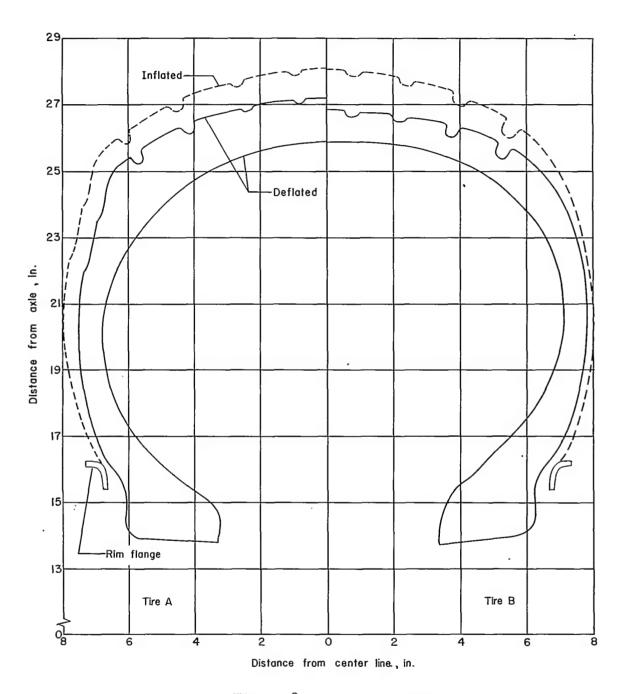


Figure 8.- Tire profiles.



(a) Approximate tread shape of tires A and B at beginning of test.





(b) At conclusion of test series C.





(c) At conclusion of test series F.





(d) At conclusion of test series I.

Tire A

Tire B

L-84905

Figure 9.- Tire wear.

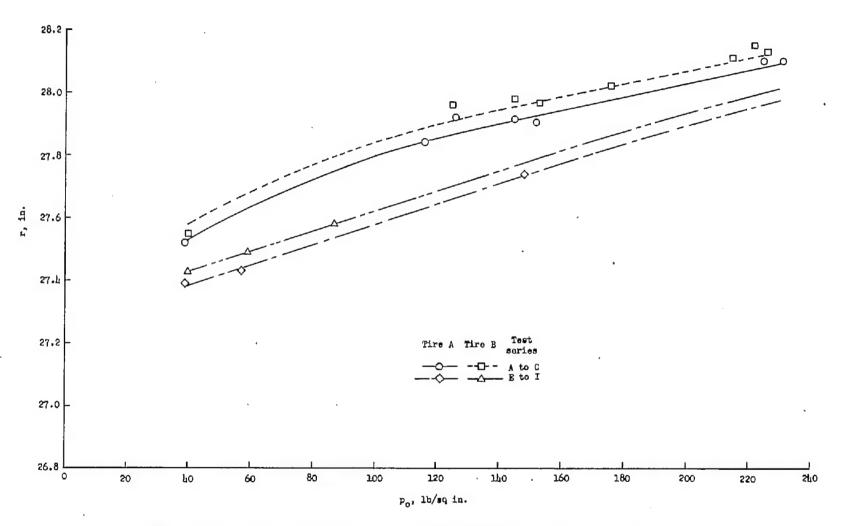


Figure 10.- Variation of unloaded tire radius with inflation pressure and tire wear. Data shown are at equilibrium and were measured after 24 hours at constant pressure.

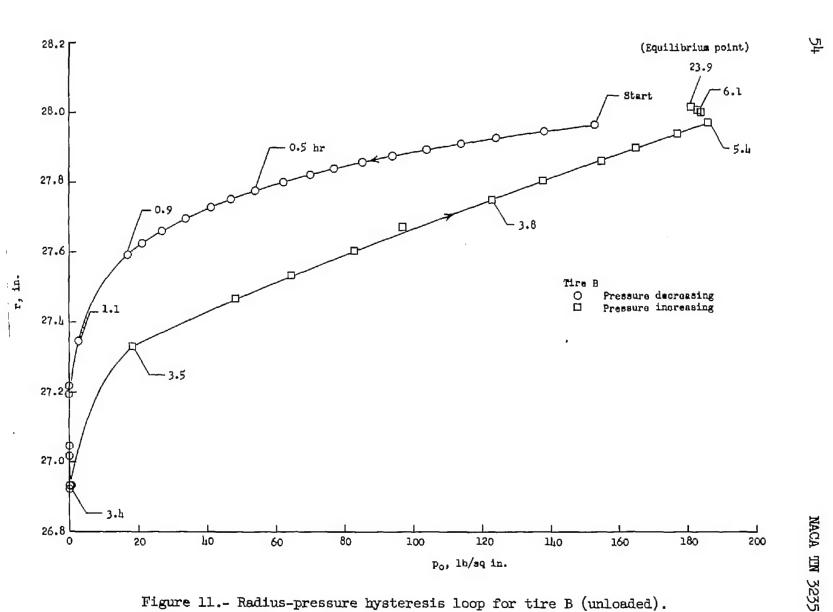


Figure 11.- Radius-pressure hysteresis loop for tire B (unloaded).

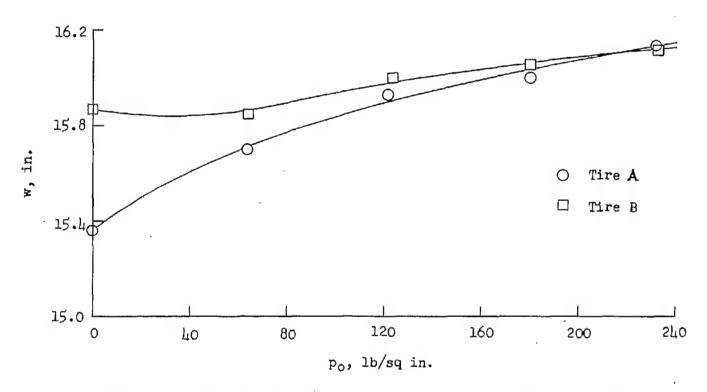


Figure 12.- Variation of maximum tire width with inflation pressure.

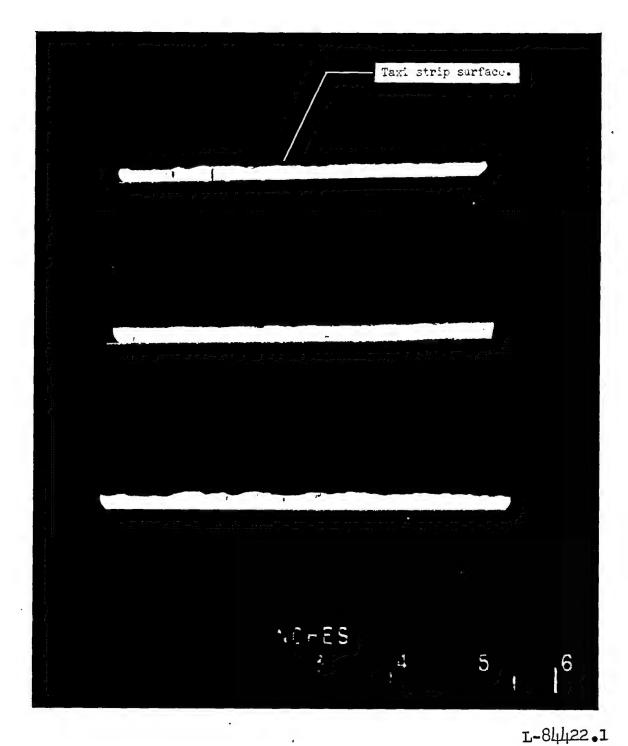


Figure 13.- Representative samples of concrete-taxi-strip surface roughness.

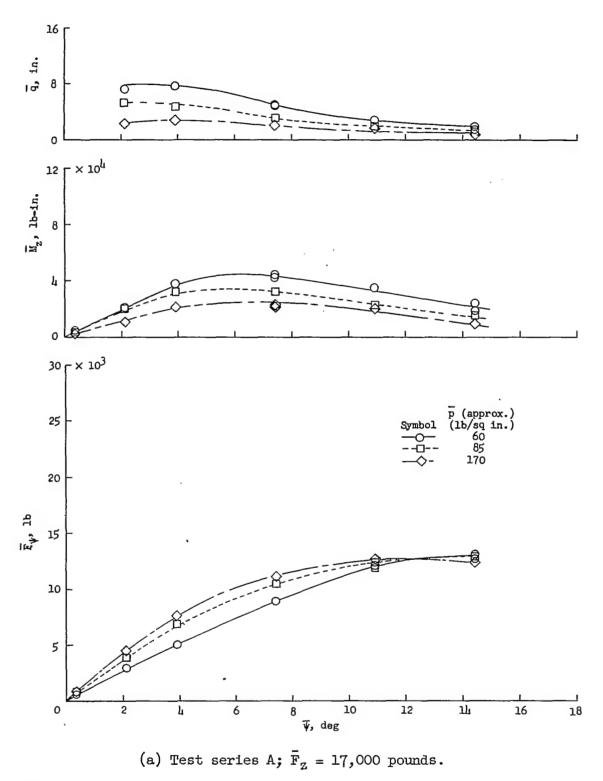
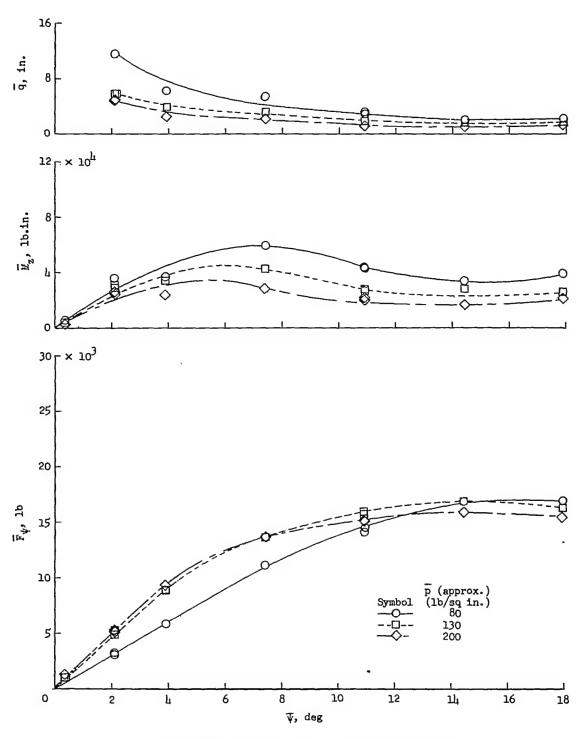
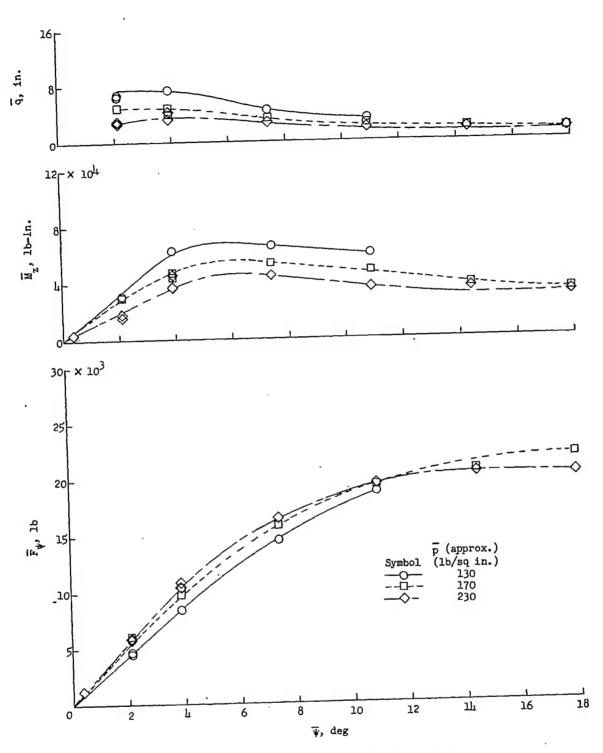


Figure 14.- Variation of normal force, self-alining torque, and pneumatic caster with yaw angle for the different vertical loads and inflation pressures investigated.



(b) Test series B;  $\bar{F}_z = 23,700$  pounds.

Figure 14.- Continued.



(c) Test series C;  $\bar{F}_z = 32,700$  pounds.

Figure 14.- Continued.

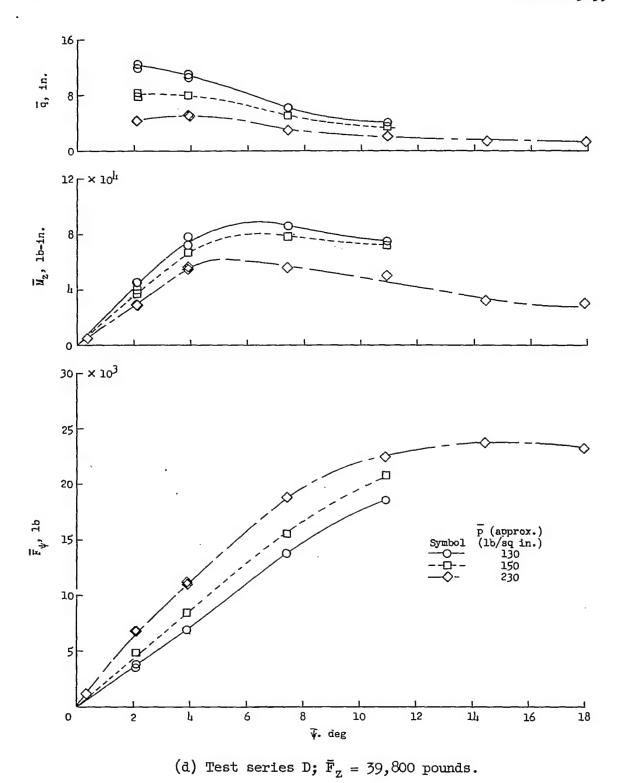
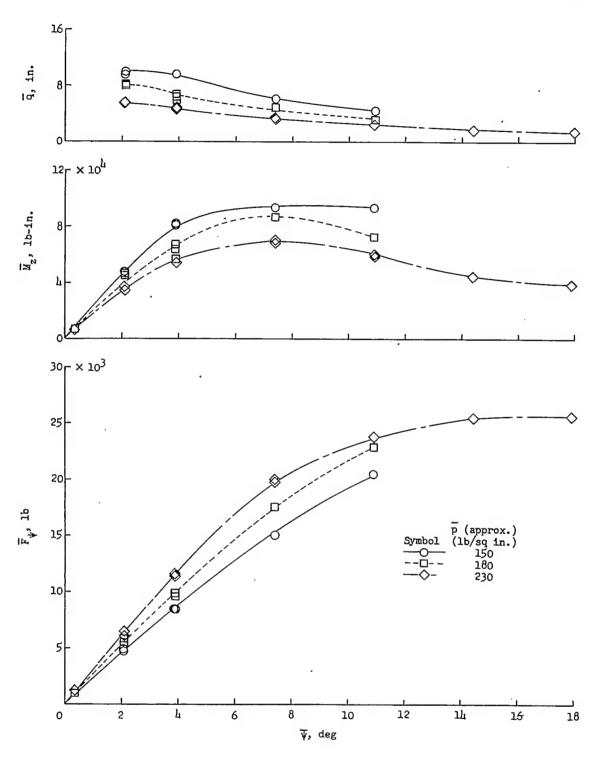
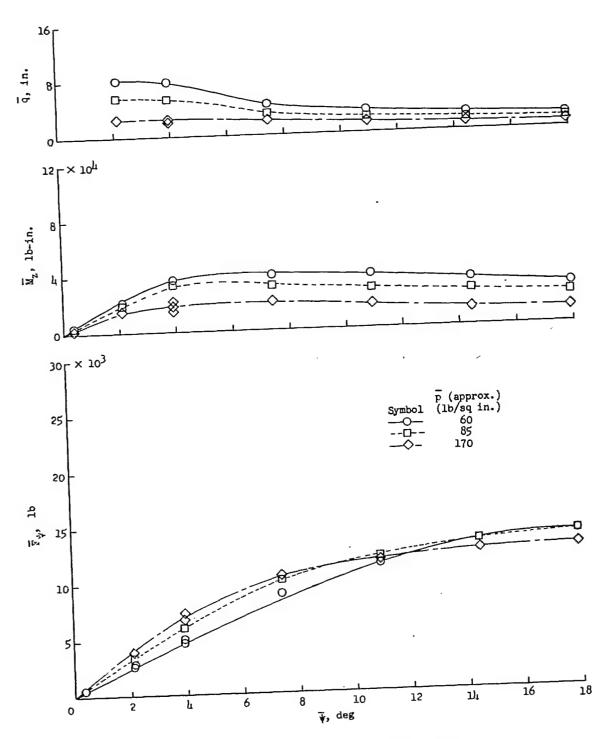


Figure 14.- Continued.



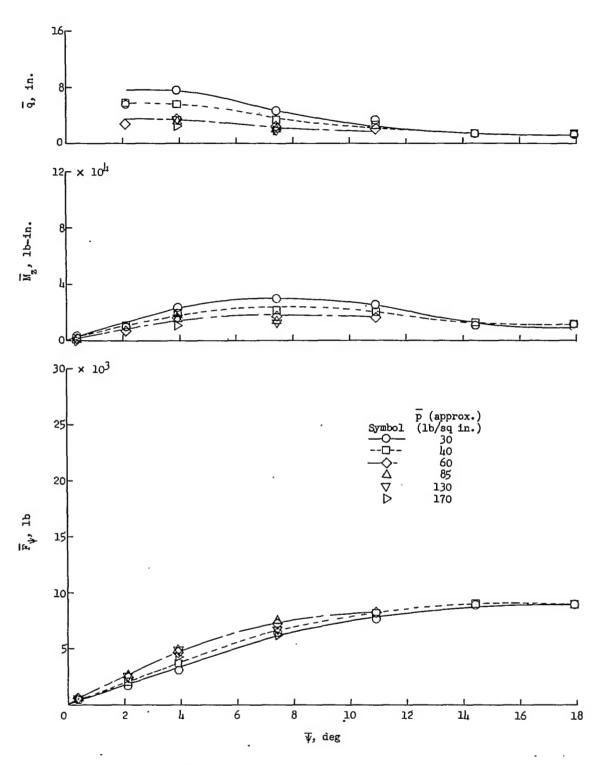
(e) Test series E;  $\vec{F}_z = 45,200$  pounds.

Figure 14.- Continued.



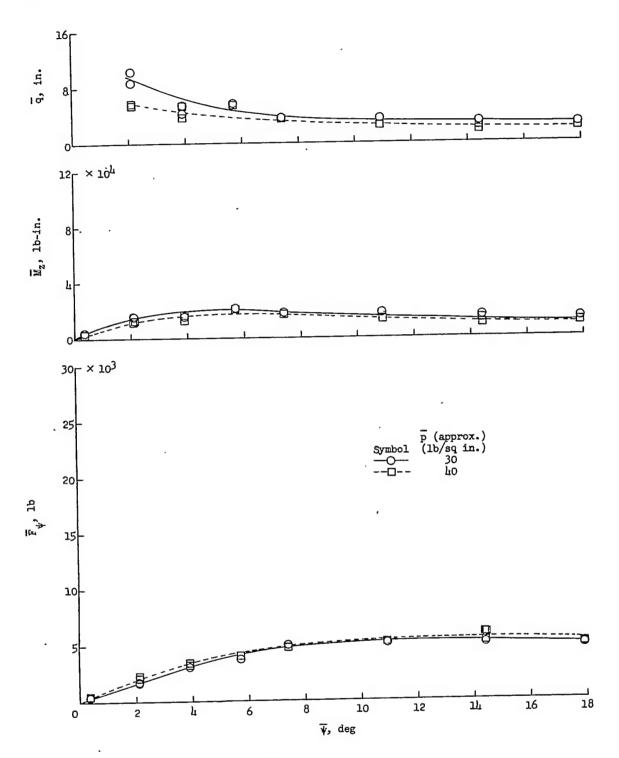
(f) Test series F;  $\bar{\mathbf{F}}_{\mathrm{Z}}$  = 17,400 pounds.

Figure 14.- Continued.



(g) Test series G;  $\bar{\mathbf{F}}_{\mathbf{Z}} = 9,600$  pounds.

Figure 14.- Continued.



(h) Test series H;  $\bar{F}_z = 9,600$  pounds. Figure 14.- Continued.

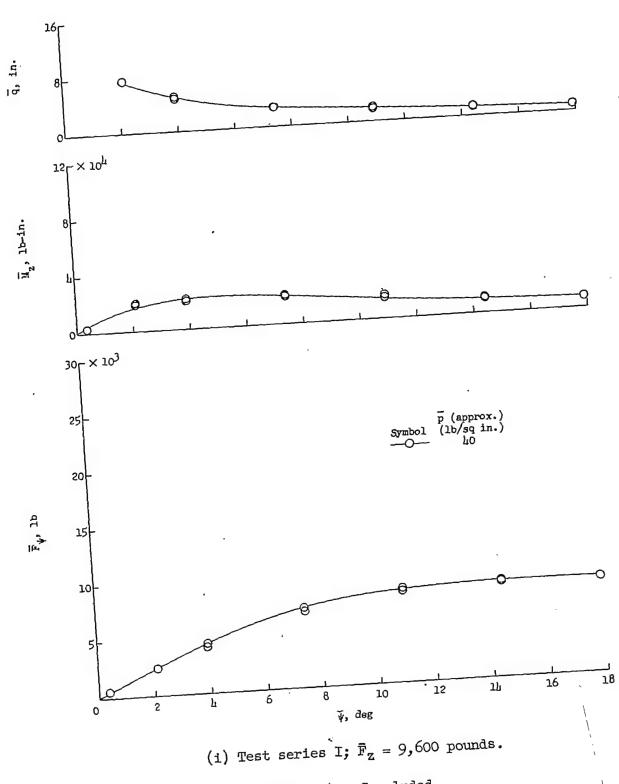
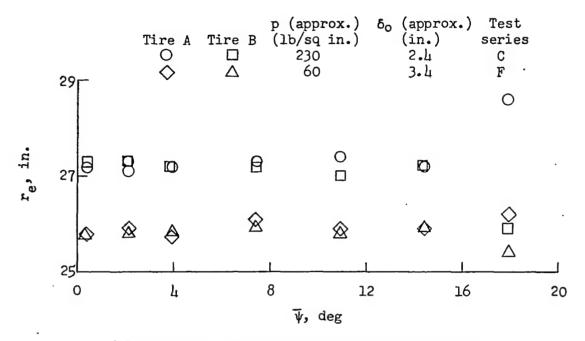
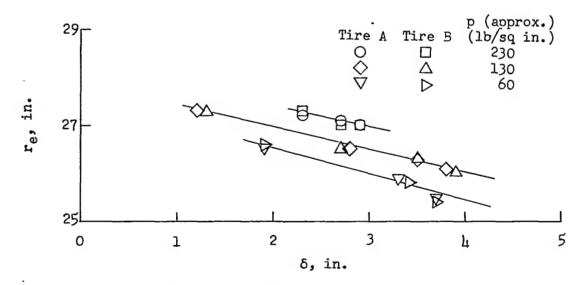


Figure 14.- Concluded.

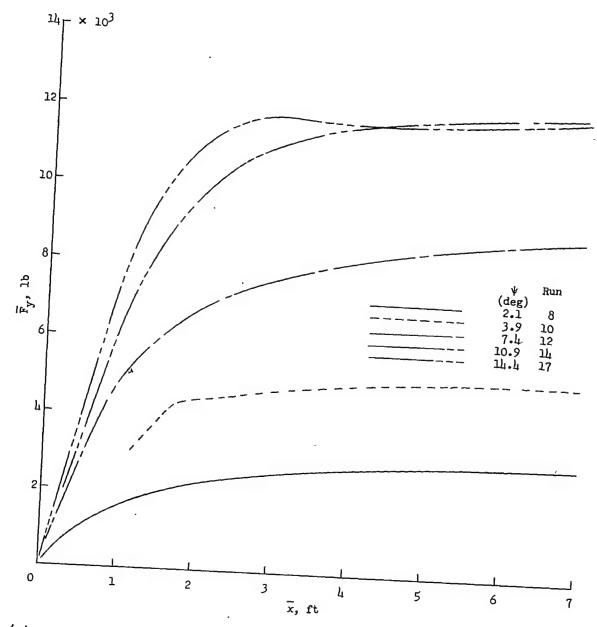


(a) Variation of rolling radius with yaw angle.



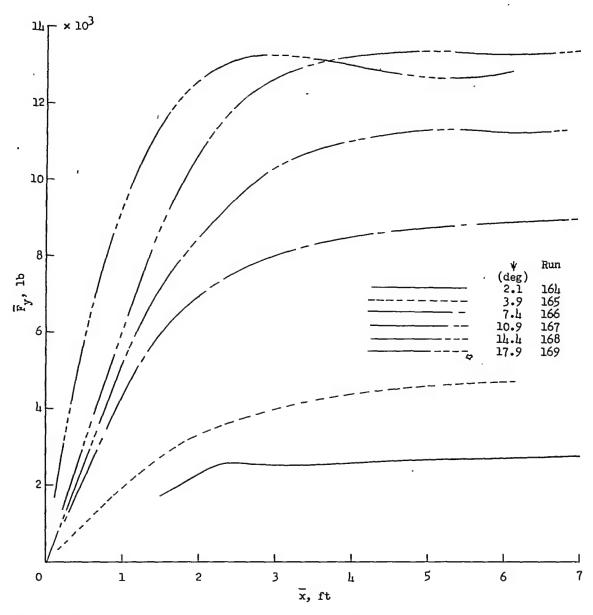
(b) Variation of rolling radius with vertical tire deflection ( $\psi = 0.35^{\circ}$  and 2.1°).

Figure 15.- Variation of rolling radius with yaw angle and vertical tire deflection.



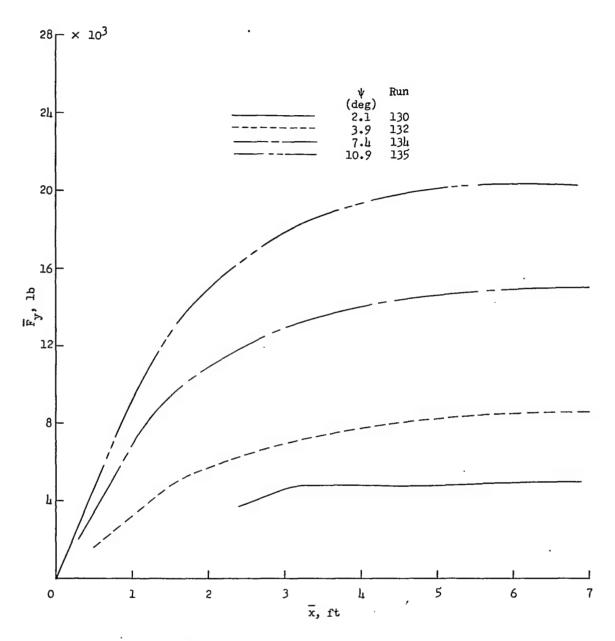
(a) Test series A:  $\vec{F}_z = 17,000$  pounds;  $\vec{p} \approx 60$  pounds per square inch;  $\vec{\delta}_0 \approx 3.7$  inches.

Figure 16.- Buildup of cornering force with distance rolled for some typical runs at several pressures and for three test series.



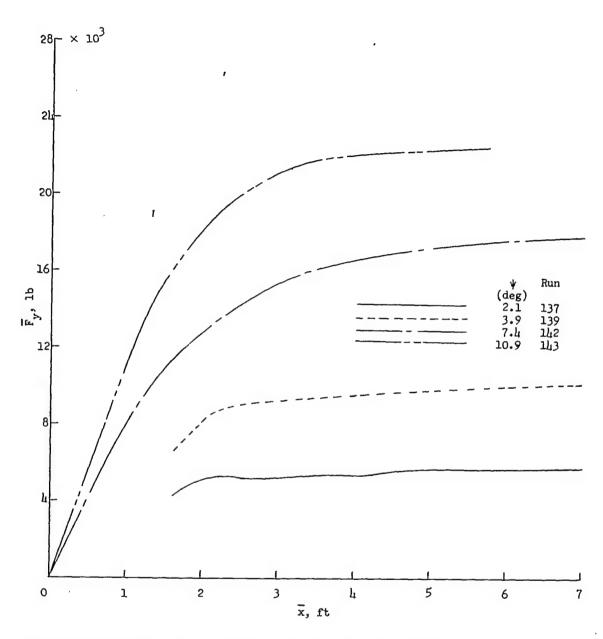
(b) Test series F:  $\bar{F}_z$  = 17,400 pounds;  $\bar{p}\approx 60$  pounds per square inch;  $\bar{\delta}_0\approx 3.5$  inches.

Figure 16.- Continued.



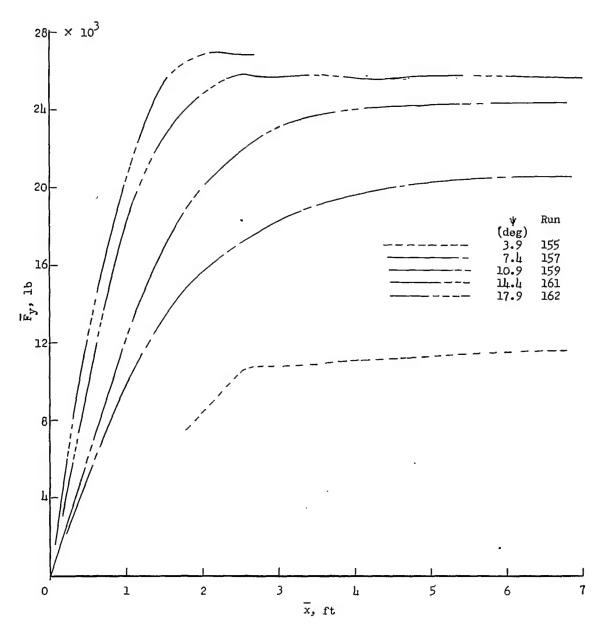
(c) Test series E:  $\bar{F}_z$  = 45,200 pounds;  $\bar{p}\approx$  150 pounds per square inch;  $\bar{\delta}_0\approx$  4.1 inches.

Figure 16.- Continued.



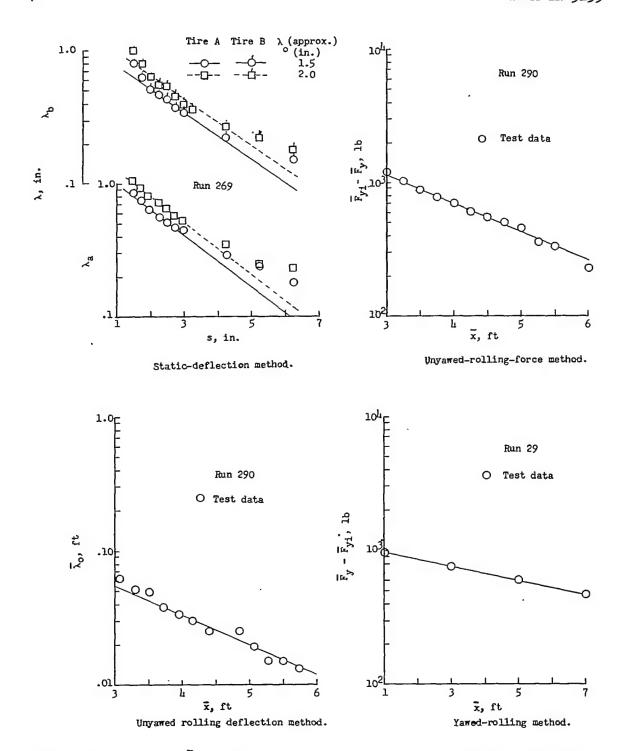
(d) Test series E:  $\bar{F}_z$  = 45,200 pounds;  $\bar{p}\approx 180$  pounds per square inch;  $\bar{\delta}_o\approx 3.6$  inches.

Figure 16.- Continued.



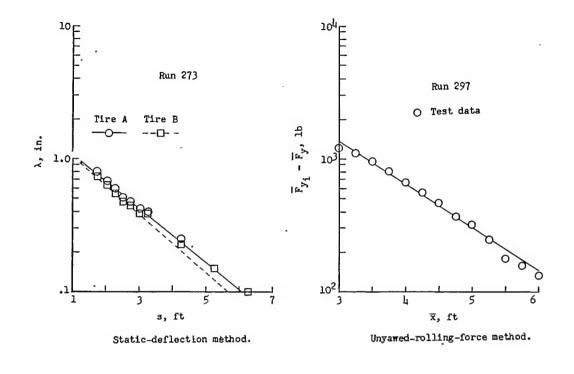
(e) Test series E:  $\bar{F}_z$  = 45,200 pounds;  $\bar{p}\approx$  230 pounds per square inch;  $\bar{\delta}_0\approx$  3.0 inches.

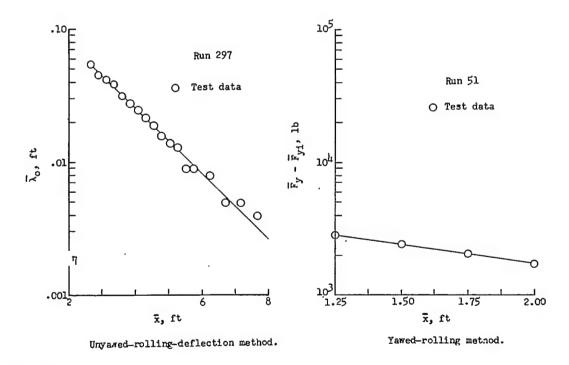
Figure 16.- Concluded.



(a) Test series A:  $\bar{F}_z$  = 17,000 pounds; for runs 269 and 290,  $\bar{p}\approx$  60 pounds per square inch and  $\bar{\delta}_0\approx$  3.7 inches; for run 29,  $\bar{p}\approx$  168 pounds per square inch and  $\bar{\delta}_0\approx$  1.8 inches.

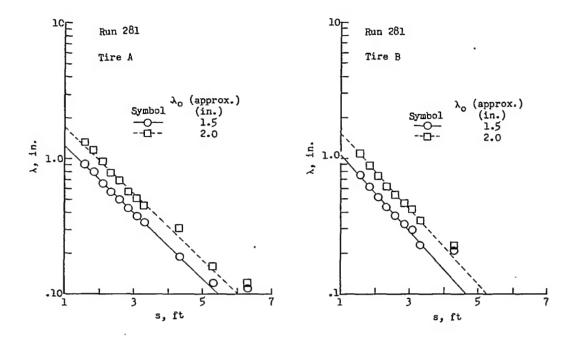
Figure 17.- Sample test data obtained from the four methods used to determine relaxation length.



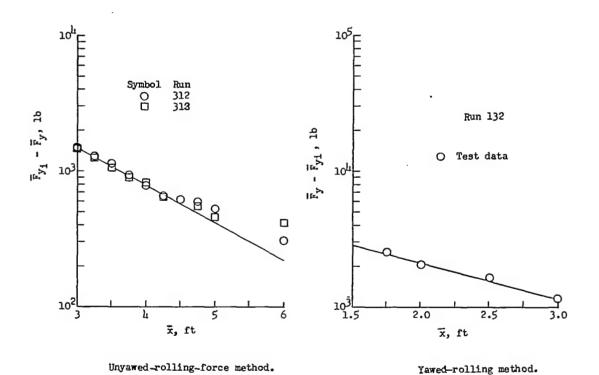


(b) Test series B:  $\bar{F}_z$  = 23,700 pounds;  $\bar{p}\approx$  130 pounds per square inch;  $\delta_o\approx$  2.8 inches.

Figure 17.- Continued.

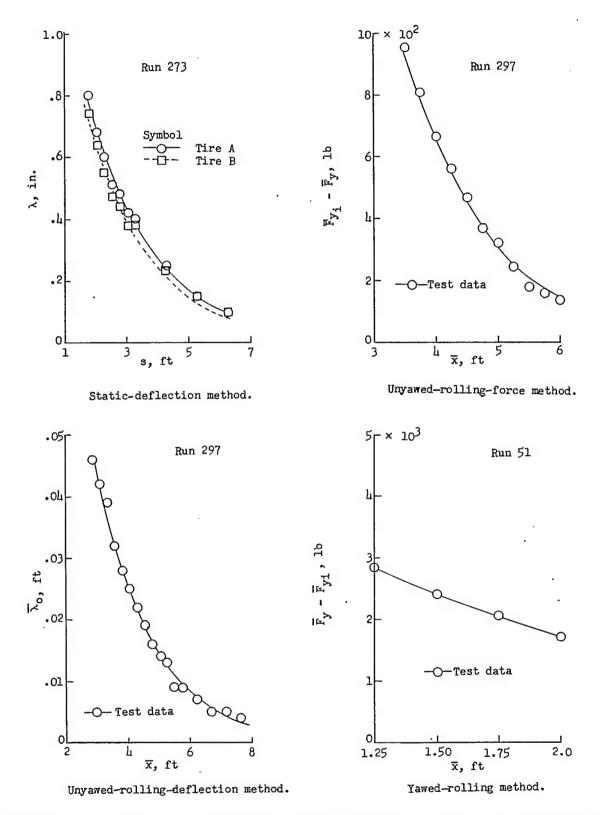


Static-deflection method.



(c) Test series E:  $\bar{F}_z$  = 45,200 pounds;  $\bar{p}$  ≈ 150 pounds per square inch;  $\bar{\delta}_o$  ≈ 4.3 inches.

Figure 17.- Continued.



(d) Data for test series B (see fig. 17(b)) replotted in linear coordinates.

Figure 17.- Concluded.

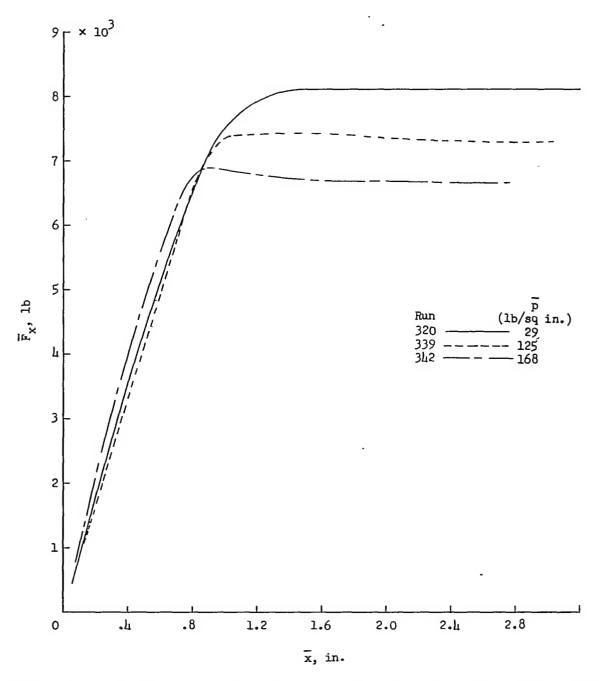


Figure 18.- Drag force buildup with horizontal distance pulled for several typical runs. Locked-wheel drag tests with  $\bar{F}_z$  = 9,730 pounds for  $\bar{F}_x$  = 0.

----

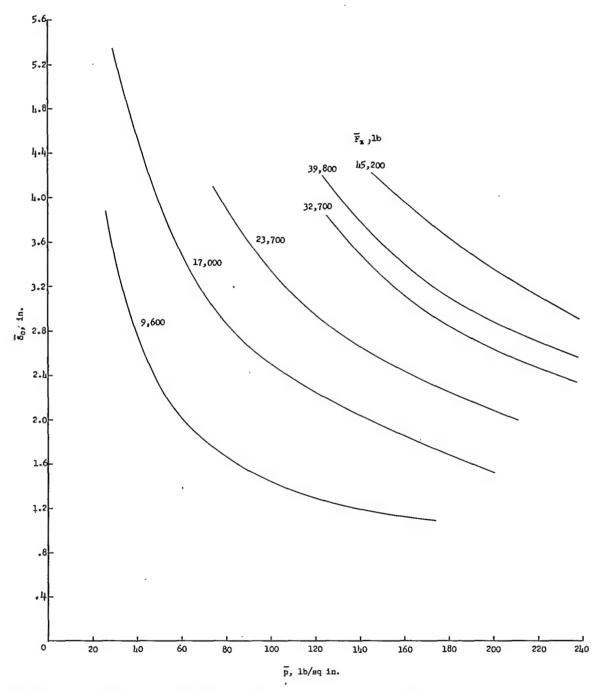


Figure 19.- Average vertical-load—vertical-tire-deflection characteristics of tires A and B.

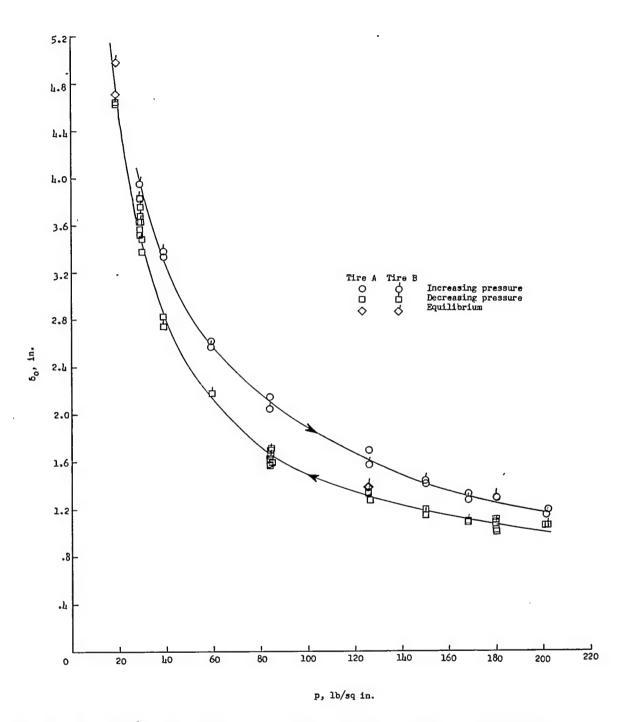


Figure 20.- Hysteresis effects on tire-pressure—vertical\_deflection relationships (table V, locked-wheel drag tests) for  $\bar{F}_Z$  = 9,730 pounds.

.....

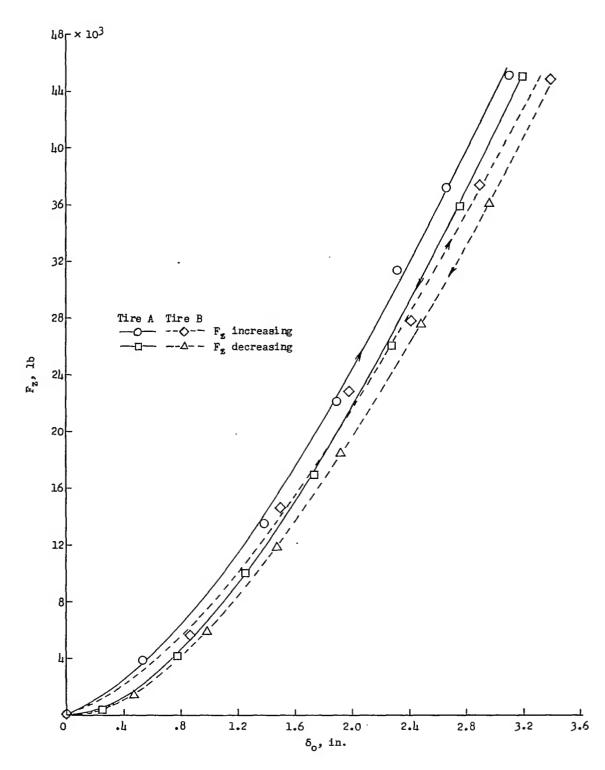


Figure 21.- Variation of vertical load with vertical tire deflection for  $p_0$  = 219 pounds per square inch. p = 221 pounds per square inch at  $F_{Z_{max}}$ .

80 NACA TN 3235

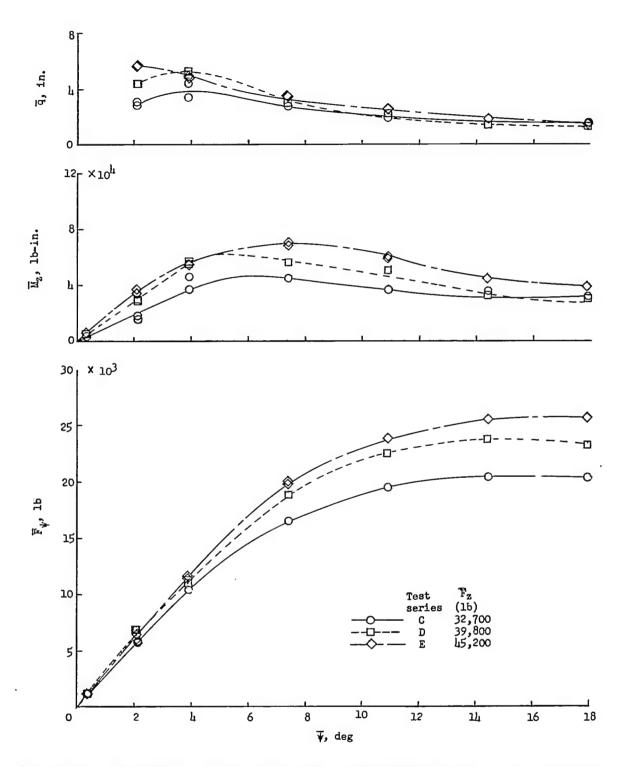


Figure 22.- Variation of normal force, self-alining torque, and pneumatic caster with yaw angle for constant inflation pressure.  $\bar{p}\approx 230$  pounds per square inch.

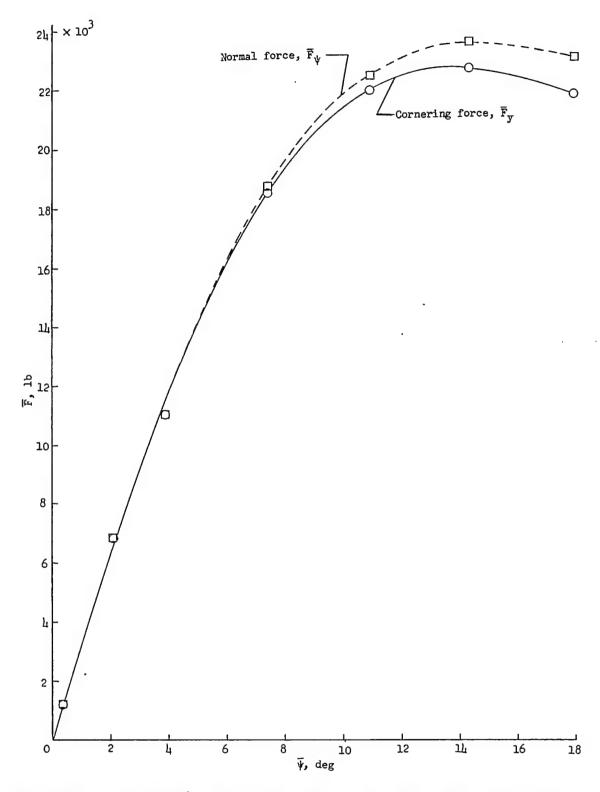
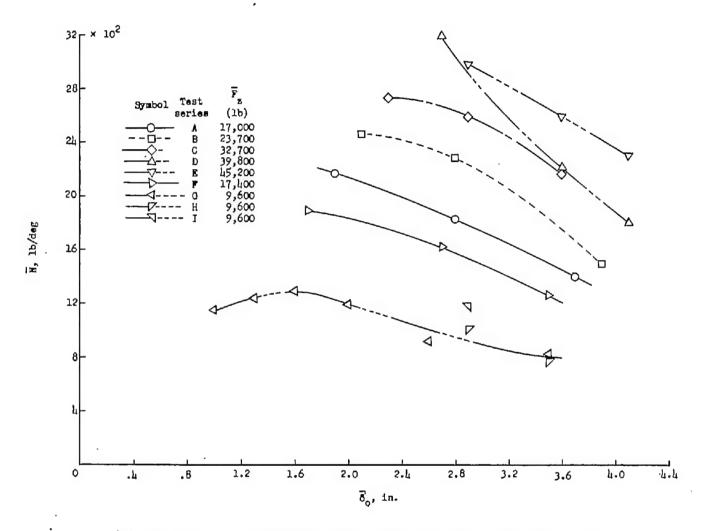
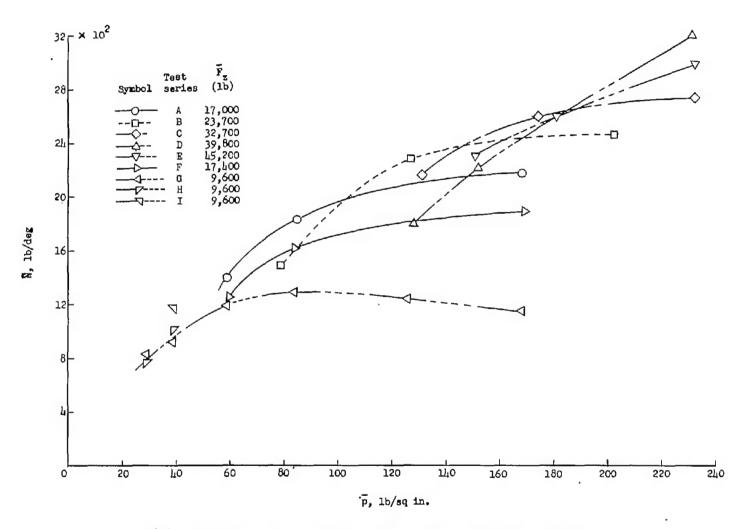


Figure 23.- Comparison of cornering-force and normal-force variations with yaw angle for a typical loading condition. (Test series D:  $\bar{F}_Z$  = 39,800 pounds;  $\bar{p}\approx$  230 pounds per square inch;  $\delta_0\approx$  2.7 inches.)



(a) Variation of cornering power with vertical tire deflection.

Figure 24.- Variation of cornering power with vertical tire deflection and inflation pressure.



(b) Variation of cornering power with inflation pressure.

Figure 24.- Concluded.



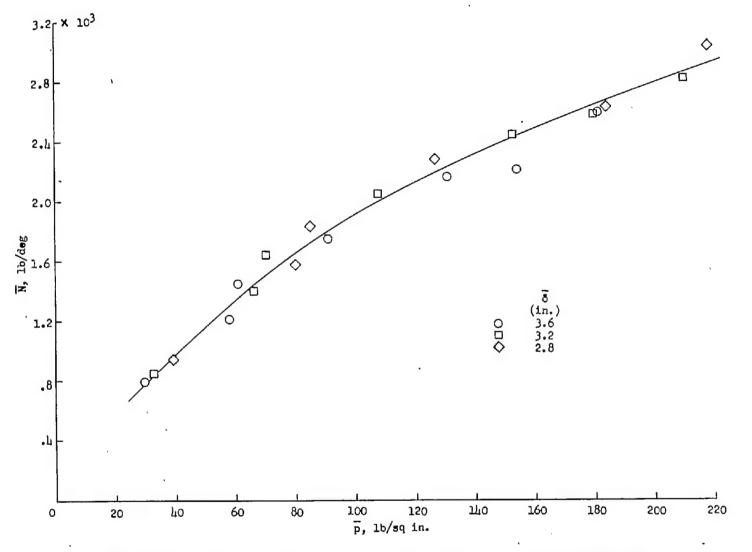


Figure 25.- Variation of cornering power with inflation pressure for several constant vertical deflections.

亚 323

•. .

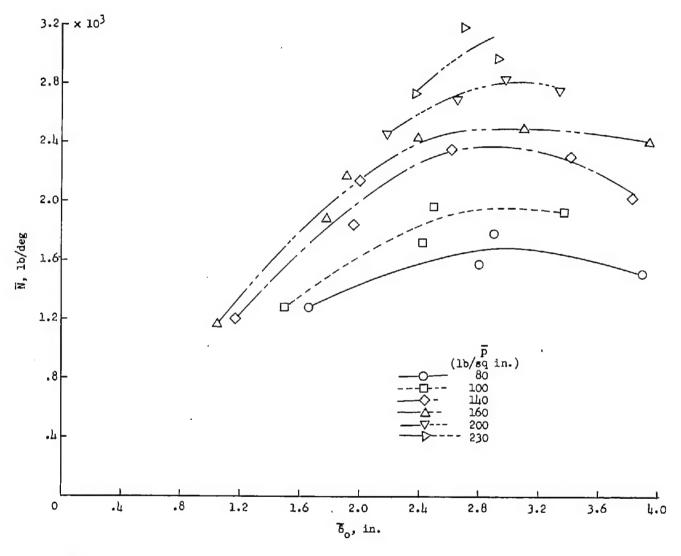


Figure 26.- Variation of cornering power with vertical tire deflection for several constant inflation pressures.

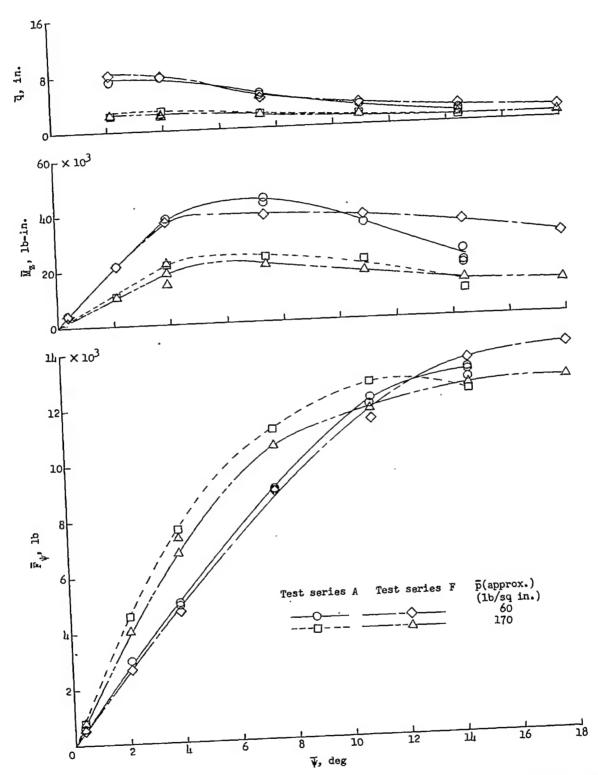


Figure 27.- Effect of tire wear on normal force, self-alining torque, and pneumatic caster for  $\bar{F}_Z \approx 17,000$  pounds.

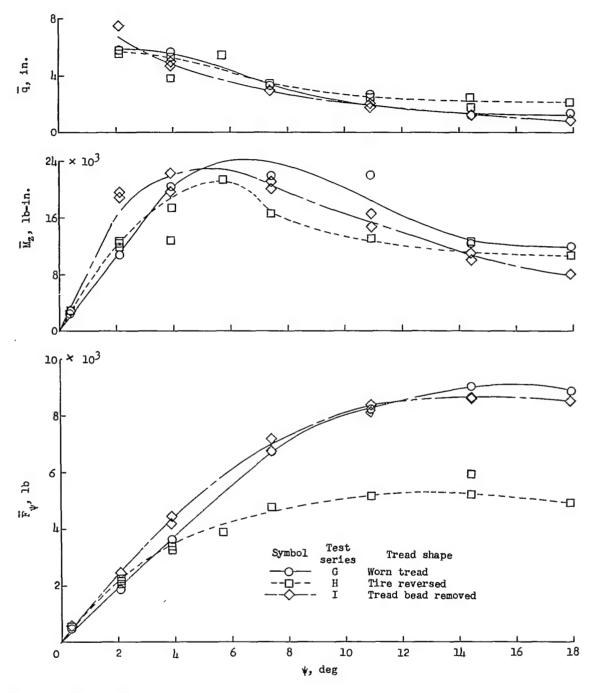


Figure 28.- Effect of tread shape on normal force, self-alining torque, and pneumatic caster for  $\bar{F}_z=9,600$  pounds and  $\bar{p}\approx 39$  pounds per square inch.

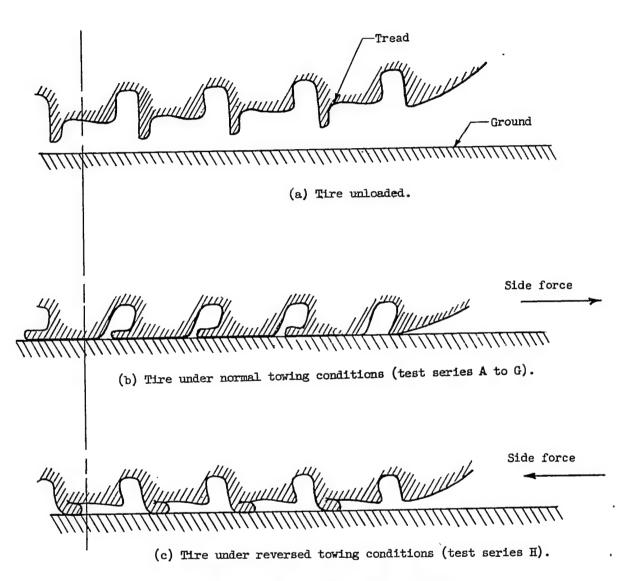
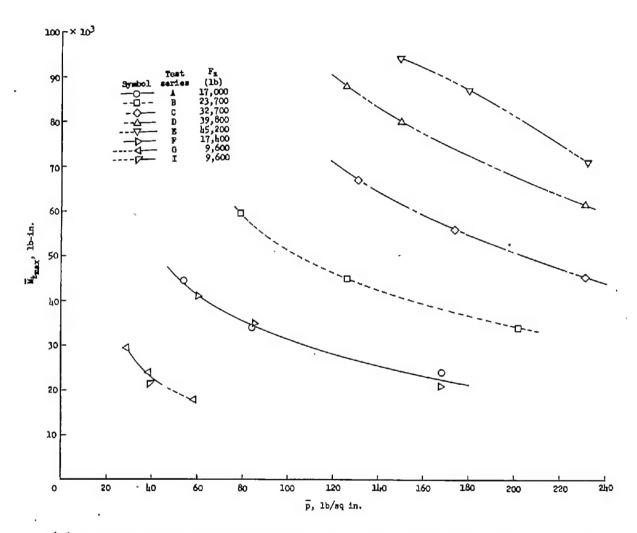
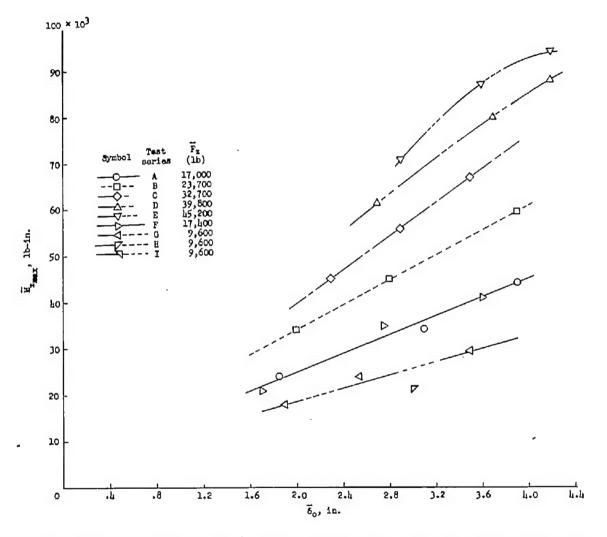


Figure 29 .- Tread bead position.



(a) Variation of maximum self-alining torque with inflation pressure.

Figure 30.- Variation of maximum self-alining torque with inflation pressure and vertical tire deflection.



(b) Variation of maximum self-alining torque with vertical tire deflection.

Figure 30.- Concluded.

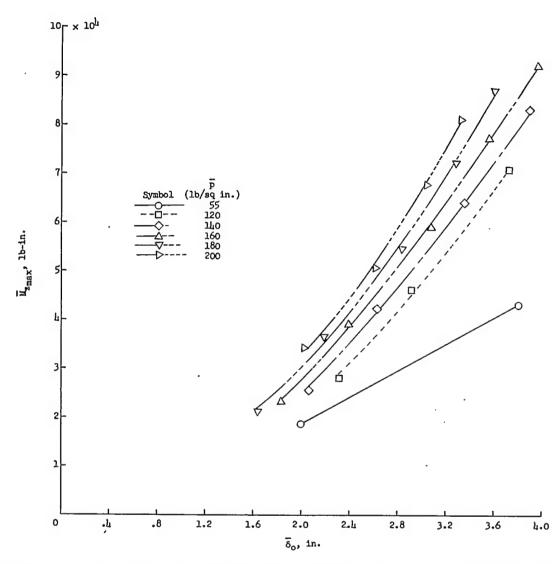


Figure 31.- Variation of maximum self-alining torque with vertical tire deflection for several constant inflation pressures.

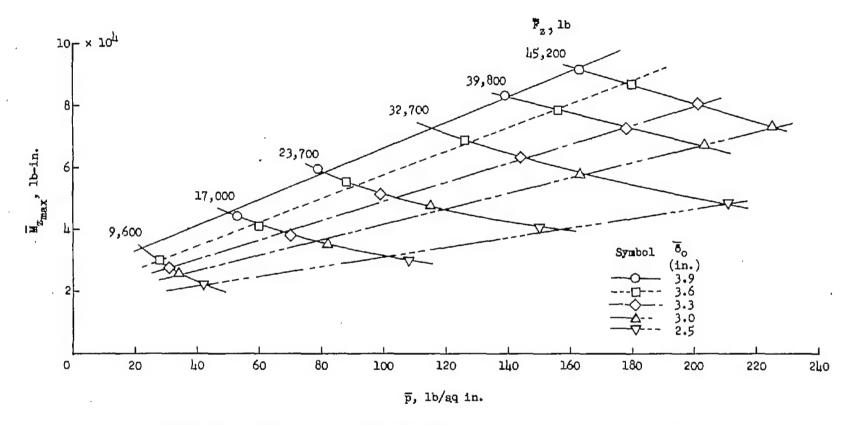


Figure 32.- Variation of maximum self-alining torque with inflation pressure for several constant vertical tire deflections.

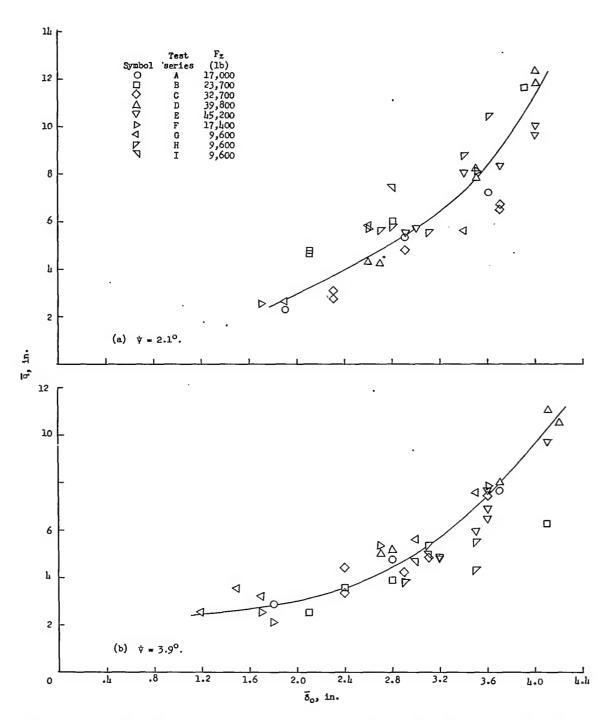


Figure 33.- Variation of pneumatic caster with vertical tire deflection for the vertical-load and yaw-angle ranges tested.

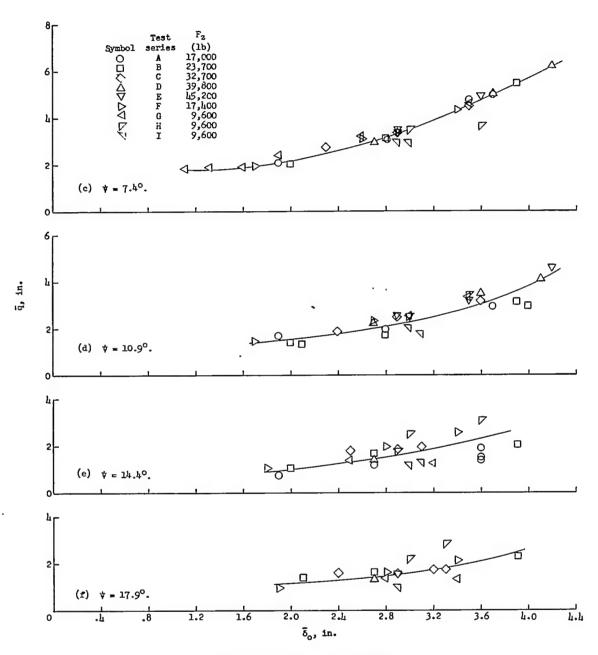


Figure 33.- Concluded.

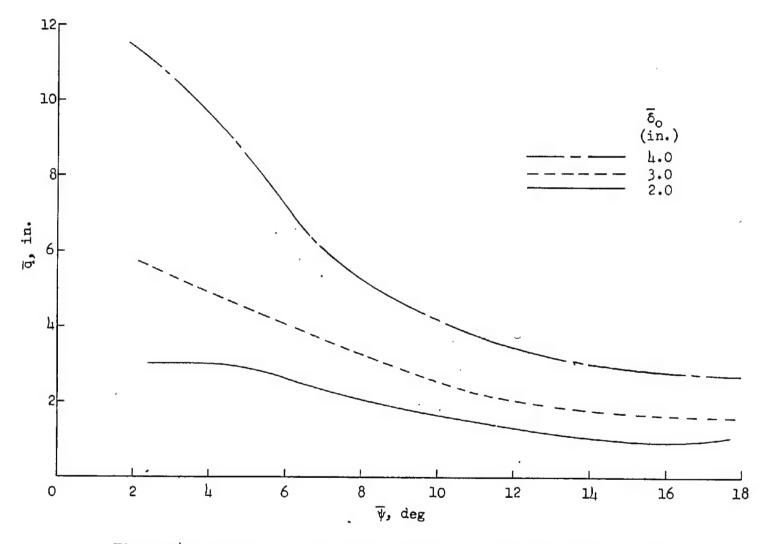


Figure 34.- Variation of pneumatic caster with yaw angle for several constant vertical tire deflections.

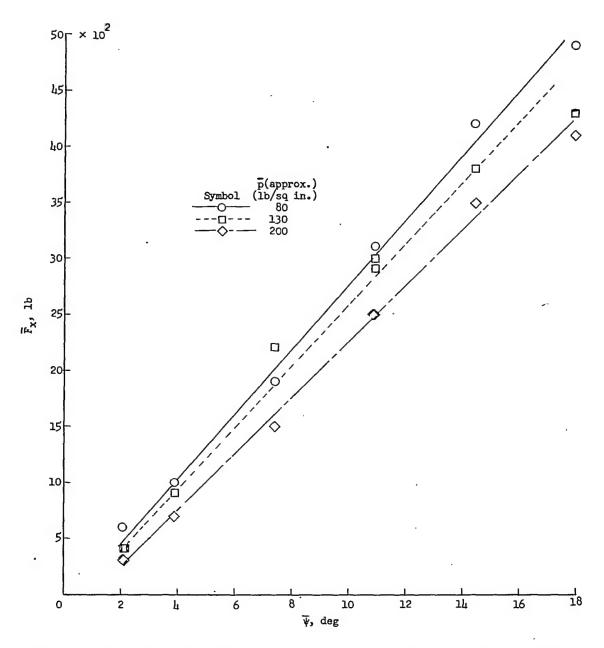


Figure 35.- Variation of drag force with yaw angle for several tire inflation pressures. (Yawed-rolling tests, test series B;  $\bar{F}_z$  = 23,700 pounds.)

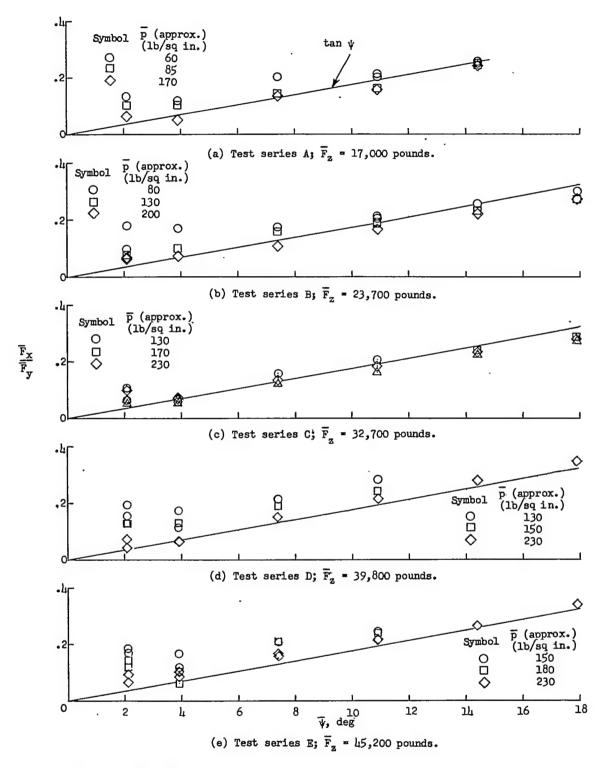


Figure 36.- Variation of the ratio of drag force to cornering force with yaw angle.

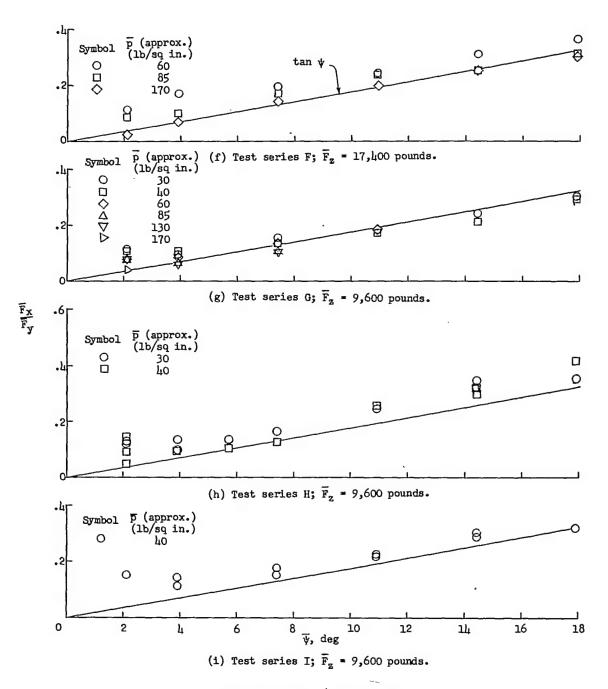
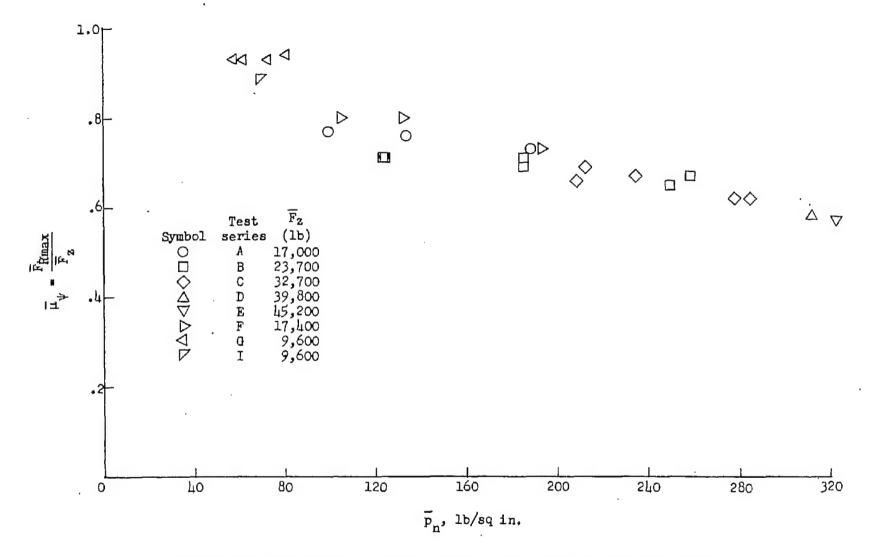


Figure 36.- Concluded.



. Figure 37.- Variation of yawed-rolling coefficient of friction with average bearing pressure.

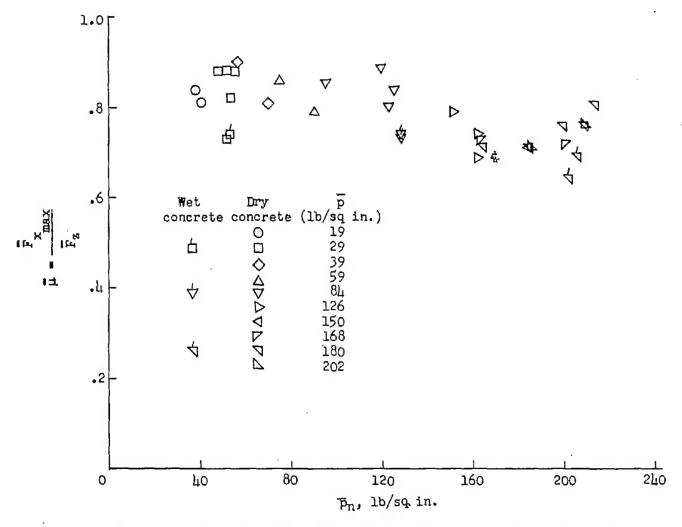
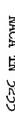


Figure 38.- Variation of sliding drag coefficient of friction with average bearing pressure.  $\bar{F}_z$  = 9,730 - 0.068 $F_{x_{max}}$  pounds.

٤

t .

t



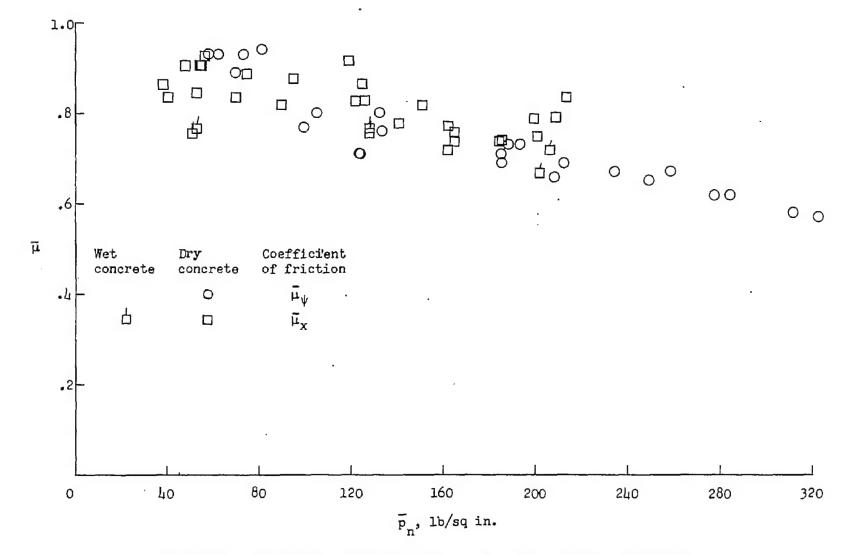


Figure 39.- Comparison of sliding drag and yawed-rolling coefficients of friction.



Figure 40.- Variation of fore-and-aft spring constant with tire inflation pressure.

NACA IIN 3235

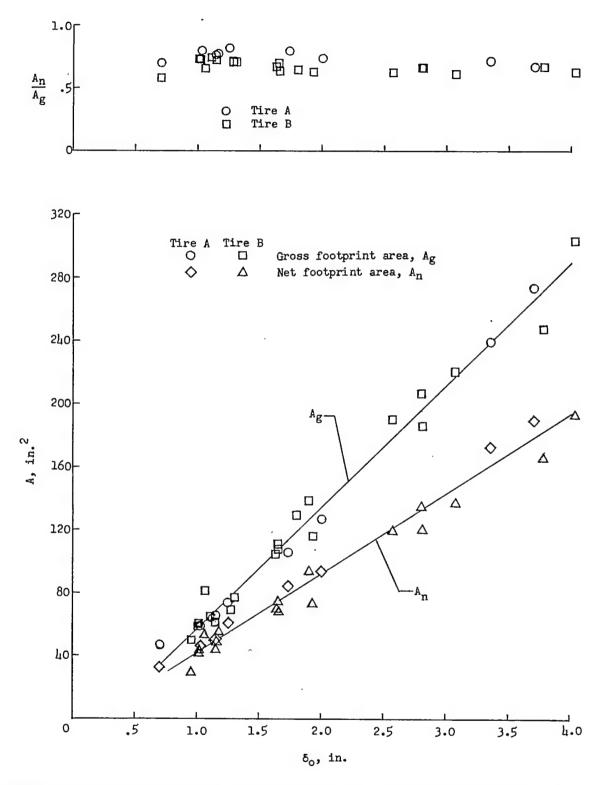


Figure 41.- Variation of gross footprint area, net footprint area, and the ratio of net footprint area to gross footprint area with vertical tire deflection for tires A and B.

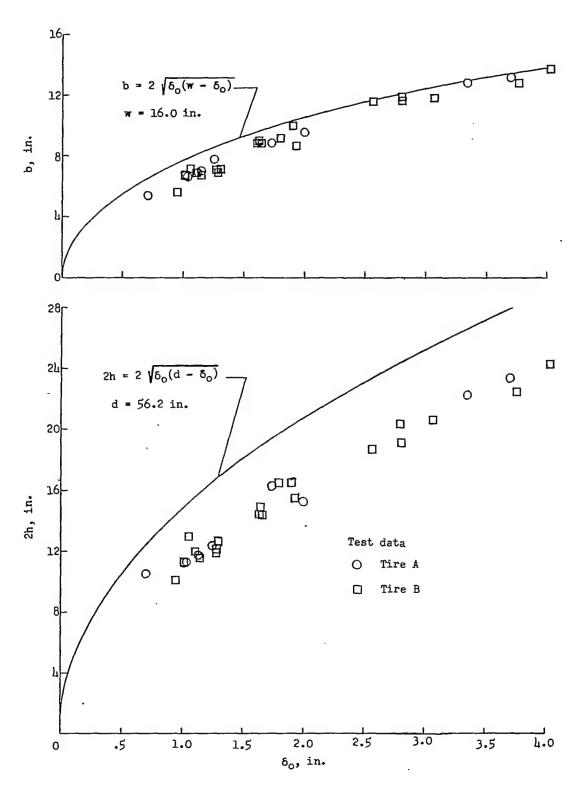


Figure 42.- Variation of footprint length and width with vertical tire deflection. (Solid lines represent chord lengths of circles having diameters equal to the diameter and width at rated inflation pressure.)

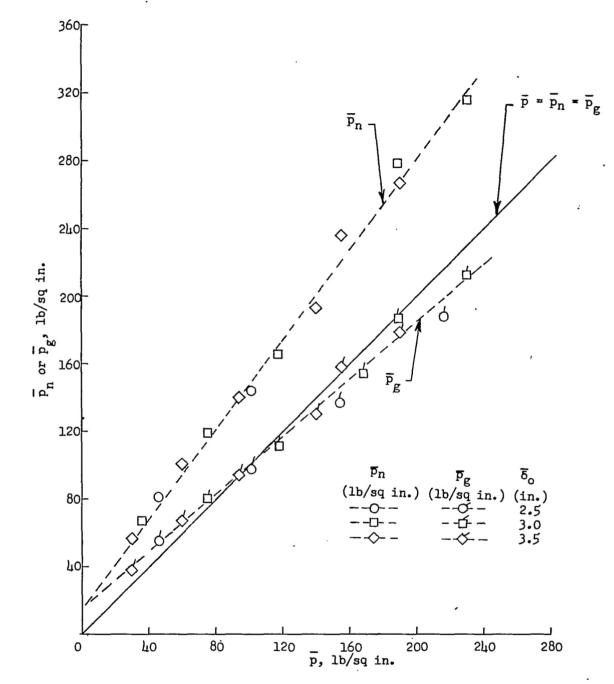
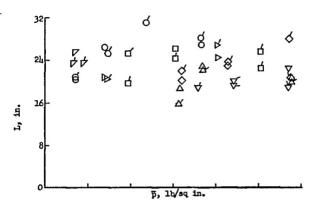
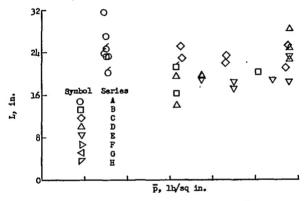


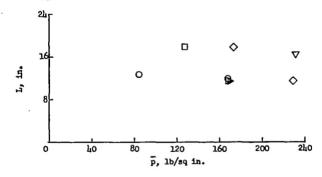
Figure 43.- Variation of average bearing and average gross footprint pressures with tire inflation pressure for several constant vertical tire deflections.



(a) Static-deflection relaxation length  $L_{\rm S}$ . Symbols without flags denote tire A values; symbols with flags, tire B values.

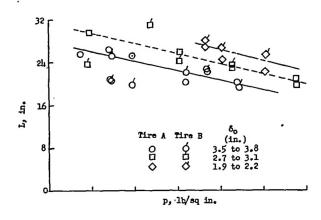


(b) Unyawed-rolling-force relaxation length  $\bar{L}_f$  and unyawed-rolling-deflection relaxation length  $\bar{L}_{\lambda}$ . Symbols without flags are for the unyawed-rolling-force relaxation length; symbols with flags, unyawed-rolling-deflection relaxation length.

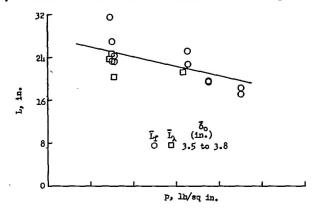


(c) Yawed-rolling relaxation length  $\bar{L}_y$ .

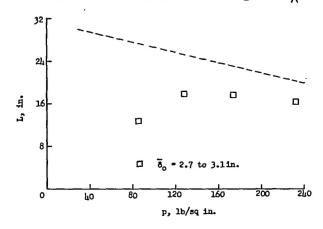
Figure 44.- Variation of the four types of relaxation length with tire inflation pressure.



(a) Static-deflection relaxation length Ls.



(b) Unyawed-rolling-force relaxation length  $\bar{L}_{\hat{\Gamma}}$  and unyawed-rolling-deflection relaxation length  $\bar{L}_{\lambda}$ .



(c) Yawed-rolling relaxation length Ly.

Figure 45.- Variation of relaxation length with tire inflation pressure for several constant vertical tire deflections.

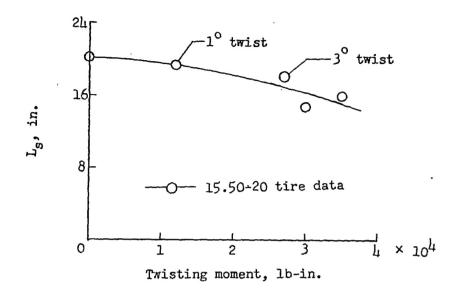


Figure 46.- Variation of static relaxation length with twisting moment or twist for a 15.50-20, type III, low-pressure tire (tire C of ref. 2).  $F_s = 3,000$  pounds;  $F_z = 20,000$  pounds;  $P_o = 81$  pounds per square inch;  $P_o = 81$  pounds per square inch.

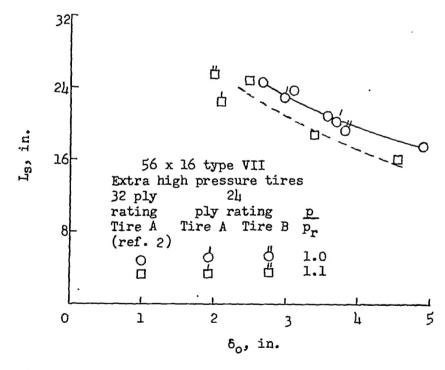


Figure 47.- Comparison of static relaxation lengths obtained from a 56 x 16, type VII, extra high pressure, 32-ply-rating tire (tire A of ref. 2) with test data.